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Study on the effects of laser shock peening on the microstructure and properties of 17-7PH stainless steel

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To investigate the surface integrity of 17-7PH stainless steel welded structural components used in aviation, laser shock peening (LSP) with different power densities was applied to stainless steel welded joints. The microstructural morphology, structural features, full-width at half-maximum, microhardness, and surface roughness of the stainless steel welded joint specimens before and after LSP were characterized and measured using SEM, TEM, XRD, a microhardness tester, and a high-resolution confocal microscope. The effects of different laser power densities on the microstructure and properties of the stainless steel welded joints were explored. Results indicate that the stainless steel welded joints exhibit a typical BCC phase. Laser shock peening promotes grain refinement in the welded joints, leading to the phase transformation of residual austenite into martensite. The surface roughness of the specimen is positively correlated with laser power density. At a power density of 5.17 GW/cm², the surface roughness increased to 1.919 μm, which is 117.08% higher than that of the non-peened specimen. The microhardness of the specimens shows a decreasing trend with increasing power density. When the power density is 2.79 GW/cm², the microhardness of the specimen significantly increases to 462.94 HV0.5, which is 22.26% higher than that of the non-peened specimen.

KEYWORDS

laser shock peening, welded joints, surface integrity, microstructure, stainless steel

1 Introduction

Precipitation-hardened (PH) stainless steels are widely used in the aerospace field due to their high strength, hardness, and corrosion resistance. They are commonly employed in welding applications for critical structural components such as engine casings and landing gear (Xu and Yu, 2008; Daymond et al., 2016; Liu et al., 2016; Ziewiec et al., 2016; Cui et al., 2022; Kugelmeier et al., 2024). In complex environments characterized by high temperatures and humidity, welded joints of structural components usually experience both high static and dynamic stresses (Li et al., 2021; Liu et al., 2023; Xiong et al., 2024). Stress corrosion cracking, fatigue vibration, and overload are major causes of structural failure (Li et al., 2014; Xiao et al., 2019; Chen et al., 2023; Jiang and Yang, 2024). To ensure the safety

of aircraft operations, it is crucial to improve the surface quality of stainless steel welded joints and enhance their surface properties.

Laser shock peening (LSP) is an effective approach to improve fatigue resistance, wear resistance, and stress corrosion resistance of metallic materials (Tsuyama, 2016; Dai et al., 2018; Sanchez et al., 2021; Shu et al., 2022; Yoo et al., 2023). Currently, researches on LSP at home and abroad mainly focus on the surface modification of materials. Zhang et al. (2020) and Zhu et al. (2020) have significantly improved the fatigue properties and surface microhardness of materials through LSP technology. Song et al. (2024) used LSP technology to increase the microhardness of the material while reducing the amount of titanium wear by more than 30%. Sundar et al. (2016) and Luet al. (2023) have demonstrated that LSP can reduce the stress corrosion cracking sensitivity of 304 stainless steel. Li et al. (2022) concluded that LSP can significantly improve the electrochemical corrosion performance of stainless steel, with the best results achieved after two impacts. However, in the field of aviation, research on the impacts of LSP on the performance of welded joints is still relatively scarce. Dhakal, Binod et al. (Dhakal and Swaroop, 2018) have discussed the significant effects of LSP on the mechanical properties and microstructure of different welded joints, concluding that LSP can be effectively used for post-weld treatment in industrial manufacturing. Rubio-Gonzalez et al. (2020) used LSP to reduce the fatigue crack propagation rate in stainless steel and improve the fracture toughness of DSS aged samples, significantly mitigating the effects of thermal aging. Zhou et al. (2022) showed that the fatigue limit of the material increased significantly from 289 MPa to 478 MPa. At the same time, LSP promoted the conversion of residual stresses in welds and heat-affected zones into residual compressive stresses of high amplitude. These studies suggest that it is technically feasible to improve the stress corrosion resistance of stainless steel welded joints by LSP. And this technology has provided a new approach for stress corrosion protection in 17-7PH stainless steel welded joints in the field of aviation.

This study investigates the impacts of LSP with optimized laser shock processing parameters on the 17-7PH stainless steel welded joints. XRD, SEM, and TEM were used to characterize and test the diffraction peaks, microstructural morphology, and structural features of the stainless steel welded joints before and after LSP. Digital microhardness tester and high-resolution confocal microscope were employed to measure the microhardness and surface roughness of the stainless steel welded joints. This research aims to study typical microstructural defects, characteristic microstructures, and microstructural evolution, and reveal the mechanisms of performance evolution in stainless steel welded joints. Compared to existing studies, the microhardness of specimens increase significantly and the grain refinement effect is better. Meanwhile, appropriate surface roughness ensures better coating application, thereby enhancing the material's corrosion resistance. The results provide new solutions for stress corrosion protection treatment in 17-7PH stainless steel welded joints for aerospace applications, offering significant scientific and theoretical value.

2 Experimental materials and methods

2.1 Experimental and sample preparation

The material used in the experiments was 17-7PH stainless steel, known for its good corrosion resistance and ductility. Its chemical composition and some mechanical properties are shown in Table 1. The 17-7PH stainless steel plates were first pickled and polished to remove the surface oxide layer. The stainless steel plates were then welded by double-sided argon arc welding, and inert gas was used to prevent oxidation of the welds. According to the welding requirements of the specimens, the welding parameters are determined by the material and dimensions of the specimens. The specific welding parameters are shown in Table 2. After welding, the front and back surfaces of the stainless steel welded joints were milled, followed by progressive polishing with 240#, 400#, 800#, and 1,200# sandpapers. Subsequently, The stainless steel welded joints were cleaned with anhydrous ethanol and dried. Finally, the dimensions of 17-7PH stainless steel welded joints shown in Figure 1 are 93 mm × 20 mm × 3 mm.

2.2 LSP process

The solid-state laser (YS1505-R200A, Xi'an Tianruida Optoelectronic Technology Co., Ltd., China) was used for the LSP, as shown in Figure 2. The laser has a pulse duration of 20 ns, a wavelength of 1,064 nm, and a spot diameter of 4 mm. The model of laser shock peening 17-7PH Stainless Steel is shown in Figure 3. Based on the extensive research conducted by the research group, optimized laser shock processing parameters were employed to strengthen the 17-7PH stainless steel welded joints. This approach aims to achieve better grain refinement effects, remove residual stresses from the material's surface, and extend the service life of the material. The optimized laser shock processing parameters are as follows: overlapping rate of 70%, pulse energies of 7 J, 10 J, and 13 J, power densities of 2.79 GW/cm², 3.98 GW/cm², and 5.17 GW/cm², and laser shock peening numbers of 3. In order to completely cover the heat-affected zone of stainless steel welded joints, the area of LSP was set to 14 mm × 14 mm. To prevent laser ablation, the transparent flowing water layer with a 2 mm thickness was used as the confinement layer and the aluminum foil with a 0.1 mm thickness was used as the laser energy absorbing layer.

2.3 Characterization and measurement methods

X-ray diffractometer (Ultima IV, Rigaku, Japan) was used to obtain diffraction patterns of the LSP areas. The scanning angle ranged from 5° to 120°, the step size was 6°/min. A field emission scanning electron microscope (FE-SEM, SU8020, HITACHI, Japan) was used to observe the microstructure of the cross-sections of the stainless steel welded joints after LSP. We employed a field emission high-resolution transmission electron

TABLE 1 Chemical composition and mechanical properties of 17-7PH stainless steel (% Mass Fraction).

C	Mn	P	S	Cr	Al	Ni	Si	HB	σ_b /MPa	σ_s /MPa	δ /%
0.09	1.0	0.035	0.03	16.0–18.0	0.75–1.5	6.5–7.75	1.0	229	1,030	380	20

TABLE 2 TIG welding parameters.

Filler material	Joint type	Nominal thickness /mm	Diameter /mm	Voltage /V	Current /A	Shield specifications (inner diameter) /mm	Argon gas flow rate /L min ⁻¹
17-7PH Stainless Steel	Butt Joint	3.2	2.4	13	140	8	6

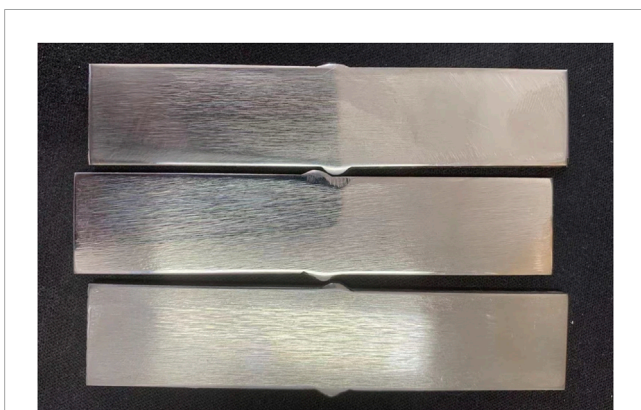


FIGURE 1 17-7PH stainless steel welded joint specimens.

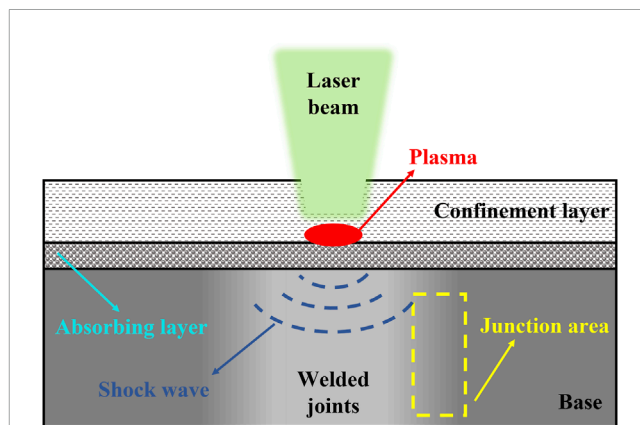


FIGURE 3 The model of laser shock peening 17-7PH stainless steel.

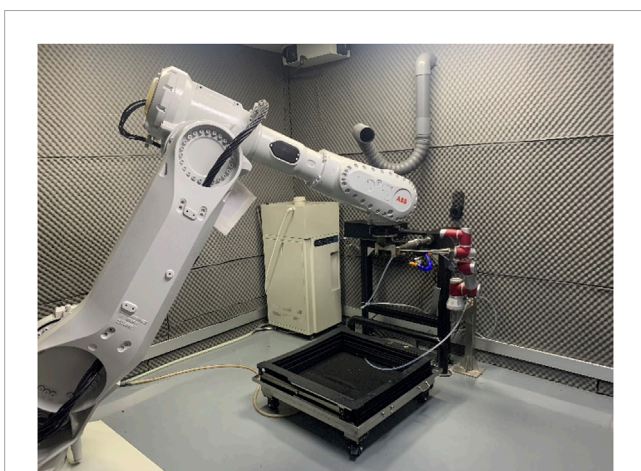


FIGURE 2 The LSP Experimental Setup.

microscope (HRTEM, Tecnai G2 F20, FEI, United States) to analyze the internal structural morphology of the welded joints. Surface topography and roughness of the stainless steel welded joints were measured using a high-resolution confocal microscope (usurf mobile, NanoFocus, Germany).

Microhardness of the surface and cross-sections of the stainless steel welded joints was measured through a digital Vickers microhardness tester (TMVS-1, Beijing Times Technology Co., Ltd., China). The loading time was 15 s with a test force of 0.5 kg. Five points were measured at depths of 100 μ m at intervals in the cross-sectional depth direction and their average values were taken as the results of the microhardness measurement.

3 Results and analysis

3.1 Microstructure

The microstructural morphology of the cross-sections of 17-7PH stainless steel welded joints before and after LSP is shown in

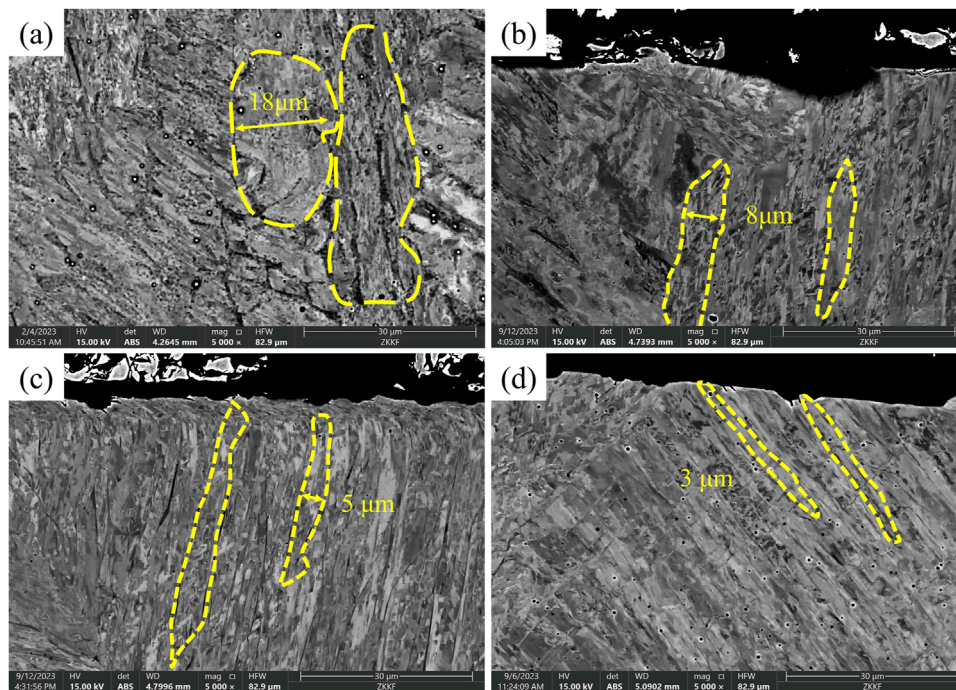


FIGURE 4 Sem Images of the welded joints before and after LSP (A) No LSP; (B) 2.79 GW/cm²; (C) 3.98 GW/cm²; (D) 5.17 GW/cm².

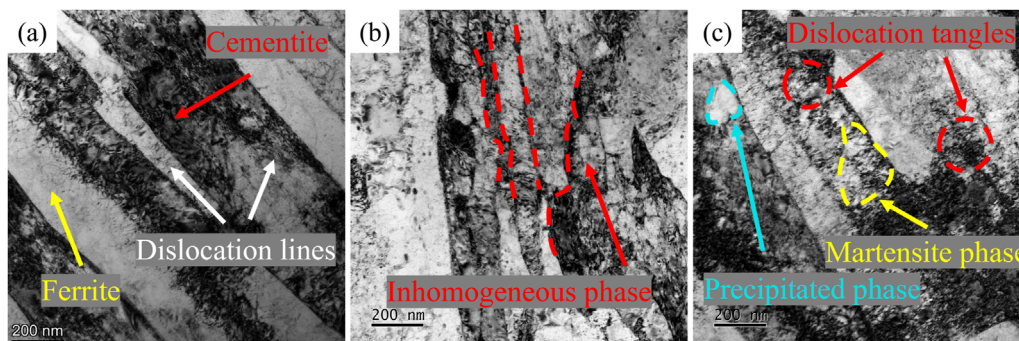
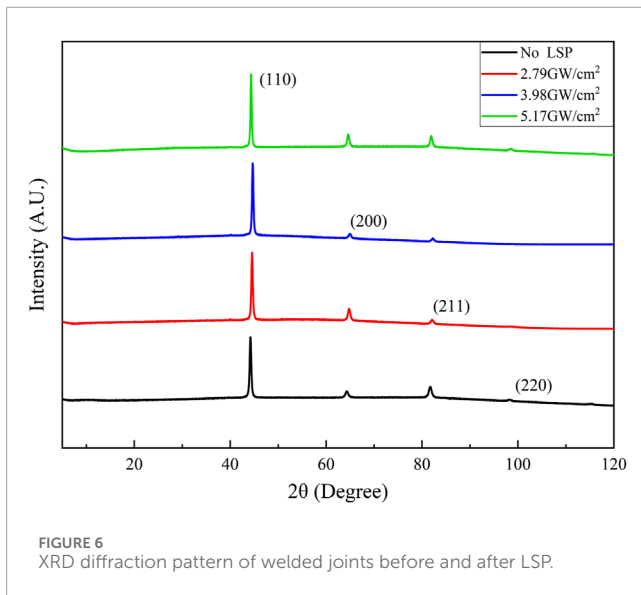


FIGURE 5 TEM Images of the welded joints before and after LSP (A) No LSP; (B) 3.98 GW/cm²; (C) 5.17 GW/cm².

Figure 4. From Figure 4A, it can be observed that the grain size at the welded joints of the stainless steel is approximately 18 μm. During the process of welding, due to uneven heat conduction and cooling rates, there are some fine weld seams present in the microstructure. The pulsed laser broke down the threshold of the material, generating a plasma and forming a burst wave. According to the Fabbro’s model (Fabbro et al., 1990; Peyre et al., 1996), the peak pressure of the shock wave can be calculated through the following Equation 1.

$$P = 0.01 \sqrt{\frac{\alpha}{\alpha + 3}} \sqrt{Z} \sqrt{I_0} \quad (1)$$

where α is the energy ratio for the gas ionization, Z is the sound impedance of 17-7PH stainless steel and confinement layer, I_0 is laser power densities, P represents the peak pressure of the shock wave (Li et al., 2024). We came to the conclusion that the peak pressure of the shock wave far exceeds the elastic yield limit of the material by calculating. Under the peening of the shock wave, slips occurred within the grains at the surface layer of the specimen, forming dislocation lines. As plastic deformation intensifies, the dislocations further evolve into subgrain boundaries. At this stage, subgrains undergo dynamic recrystallization, leading to the formation of smaller grains to achieve grain refinement. Figure 4B shows that after LSP with a laser power density of 2.79 GW/cm², the



surface grains of the stainless steel welded joints become refined, reducing the grain size to 8 μm . Figures 4C, D illustrate that as the laser power density increases, the surface grains become elongated, and grain refinement becomes more pronounced. At a power density of 5.17 GW/cm^2 , the grain size reduces to 3 μm .

Due to the minimal grain refinement observed at lower power densities, only the samples subjected to higher laser power densities were analyzed through TEM. The TEM morphology of the 17-7PH stainless steel welded joints before and after LSP is shown in Figure 5. Figure 5A reveals that the microstructure of the stainless steel welded joint consists of martensitic phases formed by elongated cementite and lamellar ferrite. Given the low carbon content in the stainless steel, the thickness of the cementite layers are approximately 150–350 nm, with some blocky cementite particles precipitating on the ferrite. Combining with the SEM morphology in Figures 4, 5B show that after laser shock peening, the cementite did not completely integrate into the austenite. Instead, the structure remains a heterogeneous mixture of ferrite and cementite. Under the influence of the LSP, an environment with high-temperature, high-pressure, and high-stress is created on the surface of the specimen. This phenomenon results in the generation of numerous dislocations and defects. During rapid cooling, the increased temperature gradient within the material creates a complex stress state, affecting the atomic arrangement of the lattice and significantly lowering the critical temperature and stability of austenite phase transformation. Concurrently, the abundance of dislocations and defects reduces the overall energy of the austenite lattice, facilitating atomic rearrangement and serving as nucleation sites for the new phase. This process promotes the transformation of residual austenite into fine-grained martensitic structure, thereby enhancing the strength and toughness of the material (Ye et al., 2012). Additionally, the high dislocation density in the stainless steel material inhibits dislocation development, and the multiple slip systems of the BCC phase lead to dislocations appearing in a cellular and tangled form (Miura et al., 2009).

3.2 Diffraction pattern analysis

The X-ray diffraction (XRD) pattern of the 17-7PH stainless steel welded joints before and after LSP is shown in Figure 6. Phase analysis reveals that the stainless steel welded joints exhibit a typical body-centered cubic (BCC) structure, specifically the α -Fe phase, with prominent peaks corresponding to the (110), (200), and (211) planes. As the laser power density increases, the intensity of the diffraction peak for the (110) plane gradually increases. This is attributed to the laser shock peening process which promotes grain refinement in the surface layer of the welded joints. Additionally, the formation of precipitated phases during the LSP process may lead to an increase in the diffraction peak intensity for the (211) plane. Compared to the non-peened samples, the laser shock peening results in significant grain refinement and lattice distortion on the material surface. The diffraction peaks tend to show increased intensity and narrowing. After high-power density laser shock peening, deformation textures appear on the surface of the stainless steel welded joints, leading to a preferred orientation of the (110) plane and an increased intensity of the diffraction peaks for this plane.

3.3 Surface integrity analysis

The three-dimensional surface morphology of the 17-7PH stainless steel welded joints before and after LSP is shown in Figure 7. From Figure 7A, it can be seen that the surface of the stainless steel welded joint, which was not subjected to laser shock peening, is relatively smooth with parallel grooves and discrete micro-peaks. These features are likely due to the micro-ploughing of the SiC particles from the sandpaper used during polishing (Duan et al., 2018). Figures 7B, C show that after LSP with lower power densities, the micro-peaks on the surface of the stainless steel welded joints gradually disappear. This is because the laser energy follows a Gaussian distribution, with the energy at the center of the spot being higher than at the edges (Li et al., 2019). This results in slight plastic deformation of the surface. Figure 7D reveals that when the laser power density increases to 5.17 GW/cm^2 , the micro-grooves on the surface of the stainless steel welded joint almost completely disappear. The relatively high laser power density leads to more pronounced plastic deformation on the surface of the welded joints.

To further characterize the impacts of Laser Shock Peening (LSP) on the surface morphology of 17-7PH stainless steel welded joints, roughness measurements were taken for the areas observed in Figure 7. The roughness was evaluated based on the arithmetic average roughness (Ra) of the profiles. The surface roughness measurement results of the stainless steel specimens are shown in Figure 8. Based on the three-dimensional surface morphology of the specimens in Figures 6, 8A indicates that the surface roughness of the 17-7PH stainless steel welded joint specimen, which were not subjected to laser shock peening, is 0.884 μm . After LSP with a power density of 2.79 GW/cm^2 , the surface roughness of the specimen increases to 1.093 μm , which is a 23.64% increase compared to the non-peened specimen. With a laser power density of 3.98 GW/cm^2 , the surface roughness reaches 1.207 μm , showing a 36.54% increase compared to the non-peened specimen.

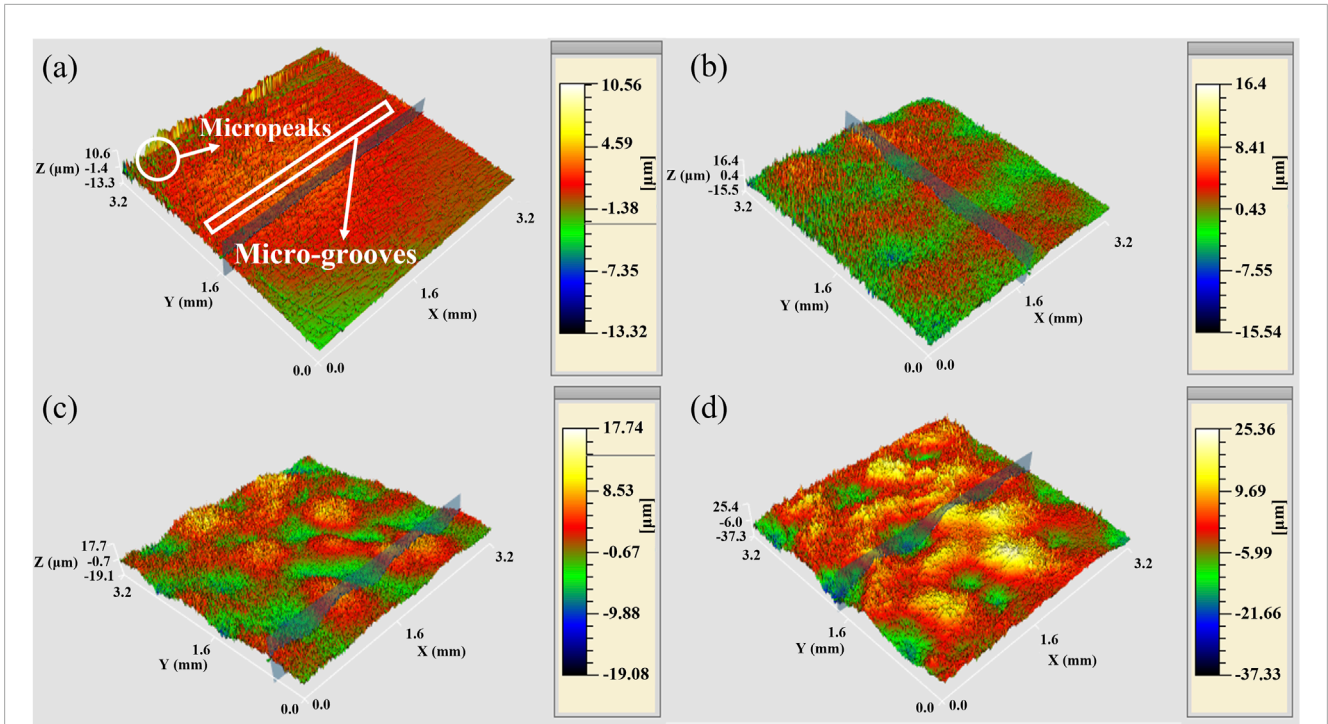


FIGURE 7 Surface 3D topography of welded joints before and after LSP (A) No LSP; (B) 2.79 GW/cm²; (C) 3.98 GW/cm²; (D) 5.17 GW/cm².

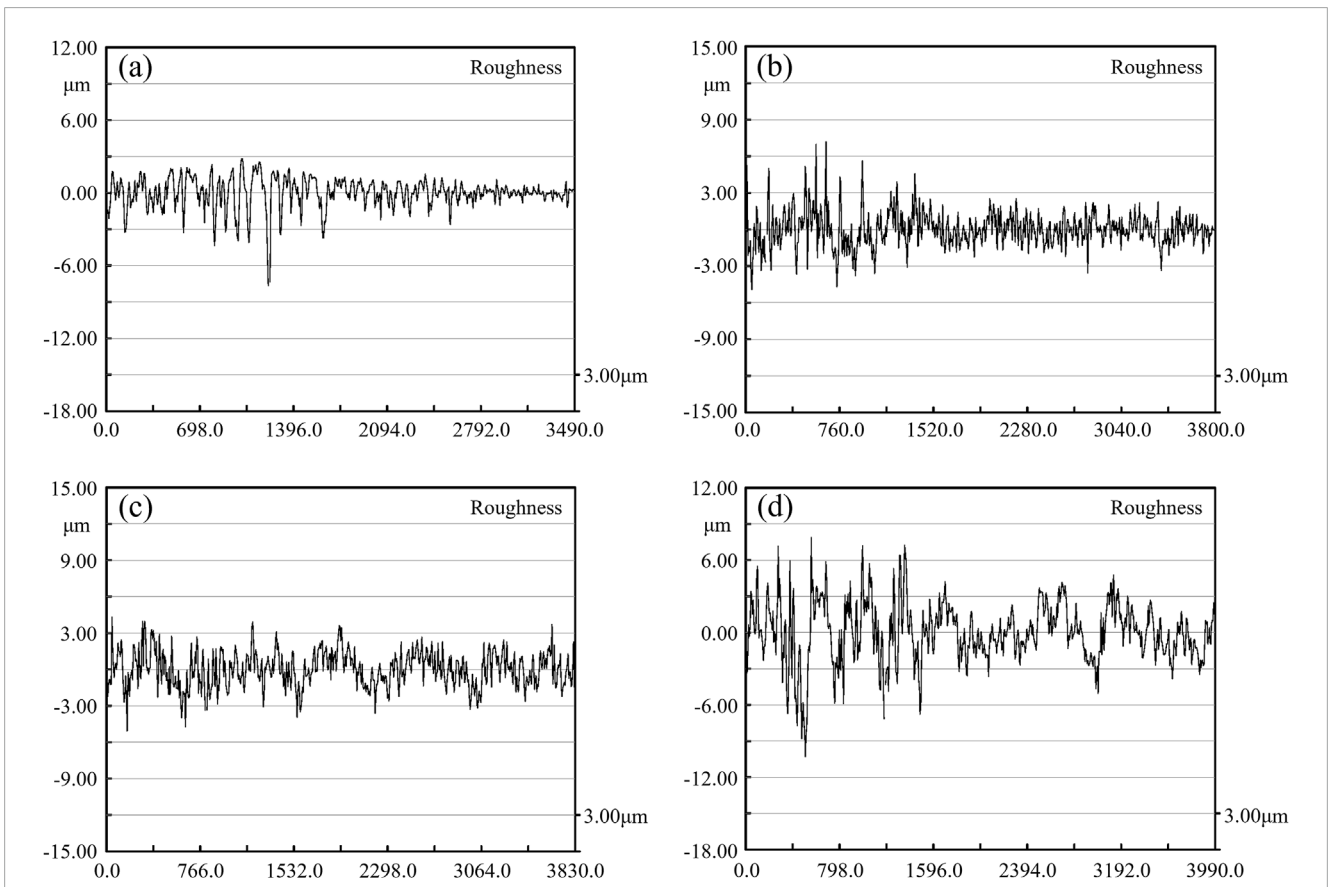
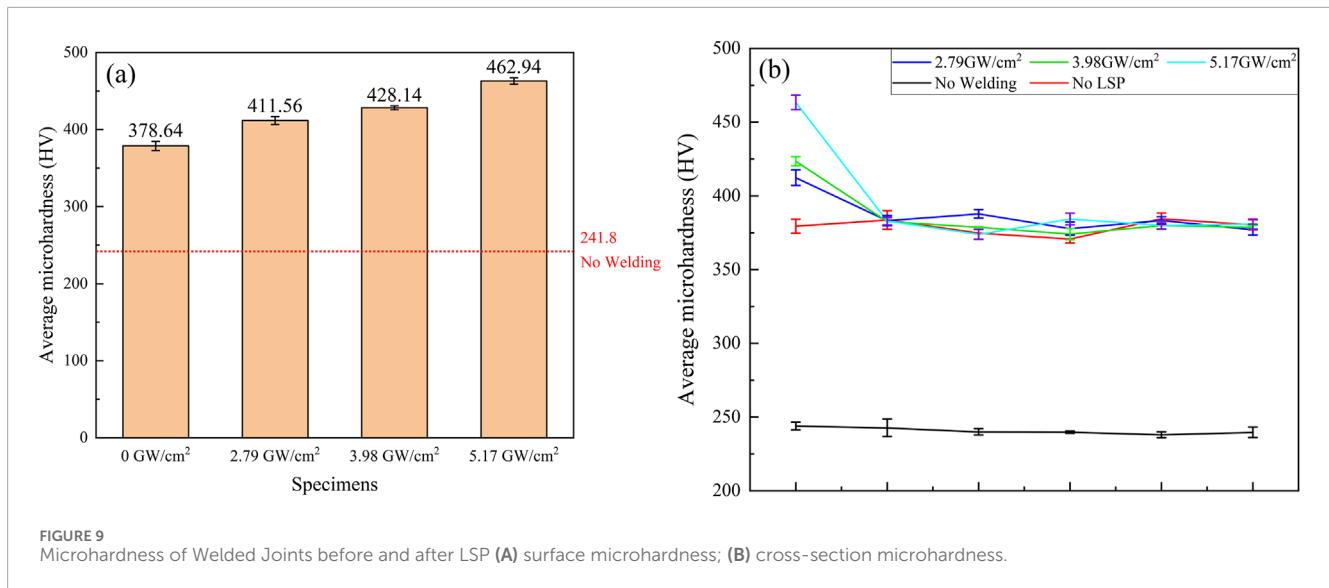


FIGURE 8 Surface roughness of welded joints before and after LSP (A) No LSP; (B) 2.79 GW/cm²; (C) 3.98 GW/cm²; (D) 5.17 GW/cm².



When the laser power density increases to 5.17 GW/cm², the surface roughness rises to 1.919 μm, reflecting a 117.08% increase compared to the non-peened specimen. Due to the Gaussian distribution of laser energy, as the laser power density increases, the surface of plastic deformation of the stainless steel welded joints become more severe, resulting in a significant increase in surface roughness.

Usually, an increase in the surface roughness of the specimen indicates the presence of more micro-grooves and cracks, which can trap moisture and other corrosive agents, creating localized galvanic environments. This can damage the oxide film on the specimen's surface, thereby reducing its corrosion resistance (Durst et al., 2008). Existing coating technologies can effectively enhance the corrosion resistance of the specimen, and an appropriate level of roughness can improve the adhesion and mechanical interlocking of the coating, providing better protection for the substrate.

3.4 Microhardness

The changes in surface and cross-sectional microhardness of the 17-7PH stainless steel welded joints before and after LSP are shown in Figure 9. Based on the microstructural morphology of the specimens in Figures 4, 9A shows that the surface microhardness of the 17-7PH stainless steel is 241.8 HV0.5. After welding, the surface microhardness increases to 378.64 HV0.5. According to the Hall–Petch Equation 2 (Luong and Hill, 2008):

$$H_v = H_{v_0} + K_{H_v} d^{-1/2} \quad (2)$$

where H_v represents the microhardness of the material after LSP, H_{v_0} is the microhardness of the base material before LSP, K_{H_v} is a constant, and d is the size of the grain at the surface of the material (Lu et al., 2010). Dislocations are the primary defects responsible for plastic deformation in metallic materials, an increase in grain boundaries following grain refinement enhances the interactions between dislocations (Ma et al., 2020). The presence of more grain boundaries obstructs dislocation movement, thus strengthening

the material's ability to withstand external forces. Combined with the Hall–Petch equation, grain refinement leads to an increase in material microhardness. With increasing laser power density, the surface grains of the stainless steel welded joints become progressively finer, resulting in an increase in surface microhardness. When the laser power density reaches 5.17 GW/cm², the surface microhardness of the stainless steel welded joint increases to 462.94 HV0.5, which is a 22.26% improvement compared to the microhardness of the welded joint before LSP. From Figure 9B, it can be seen that the depth of the grain refinement layer caused by laser shock is only a few tens of micrometers. The hardness of the stainless steel welded joint cross-section gradually decreases to the hardness level of the untreated state as the depth increases.

Additionally, we observed the martensite-ferrite phase in Figure 5. Combined with the research of P. V. Yasni et al. (2010), we concluded that the surface of the stainless steel specimen exhibits refined lamellar martensite after LSP, with an increased length of subgrain boundaries. This results in enhanced fracture energy strength and significantly improves the impact toughness of the specimen.

4 Conclusion

- 1) After laser shock peening, the surface grains of the 17-7PH stainless steel welded joints become refined, with grain size decreasing as the laser power density increases. At a power density of 5.17 GW/cm², the grain size reduces to 2–5 μm, and the depth of the laser shock peening-affected layer is approximately 50 μm. The residual austenite inside the weld joint undergoes phase transformation to martensite, and cellular and tangled dislocations form between the grains.
- 2) Due to the Gaussian distribution of laser energy, as the laser power density increases, the plastic deformation on the surface of the stainless steel welded joints intensifies, leading to an increase in surface roughness. At a laser power density of

5.17 GW/cm², the surface roughness of the stainless steel welded joint rises to 1.919 μm, which is a 117.08% increase compared to the non-peened specimen.

- 3) The surface microhardness of the 17-7PH stainless steel welded joints is 378.64 HV0.5, significantly higher than the base material microhardness of 241.8 HV0.5. The surface microhardness increases positively with laser power density. At a power density of 5.17 GW/cm², the surface microhardness of the stainless steel welded joint reaches 462.94 HV0.5, which is a 22.26% increase compared to the non-peened specimen.
- 4) Laser shock peening significantly improves the performance of critical components such as aircraft landing gear, reducing maintenance costs while enhancing overall aircraft safety. However, current laser shock peening has stringent geometric requirements for the parts, which can lead to uneven strengthening effects.

Data availability statement

The original contributions presented in the study are included in the article/Supplementary Material, further inquiries can be directed to the corresponding authors.

Author contributions

KL: Conceptualization, Investigation, Methodology, Validation, Writing—original draft, Writing—review and editing. WY: Funding acquisition, Validation, Writing—review and editing. YL: Resources, Visualization, Writing—review and editing. HB: Conceptualization, Writing—review and editing. YC: Formal Analysis, Supervision, Validation, Writing—review and editing. YW: Project administration, Writing—review and editing.

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Conflict of interest

Authors WY and YL were employed by Comac Shanghai Aircraft Manufacturing Co., Ltd.

The remaining authors declare that the research was conducted in the absence of any commercial or financial relationships that could be construed as a potential conflict of interest.

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Supplementary material

The Supplementary Material for this article can be found online at: <https://www.frontiersin.org/articles/10.3389/fmats.2024.1484698/full#supplementary-material>

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