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# Optimal real-time tuning of autonomous distributed power systems using modern techniques

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This work considers using a novel heuristic population-based evolutionary algorithm [viz., the moth flame optimization (MFO) algorithm] to regulate the conventional controller installed in an autonomous power system (APS). The moth flame optimization algorithm intends to produce the optimal magnitudes of the proportional-integral-derivative plus second derivative (PIDD<sup>2</sup>) controller parameters along with its first- and second-order low-pass filter constraints (installed in the investigated autonomous power system). The present task includes a comparison of the voltage response profiles of the investigated system obtained by the proposed moth flame optimization-based proportional-integral-derivative plus second derivative controller and those obtained by other algorithms (conveyed in current state-of-the-art literature) based on a proportional-integral controller. A fast-acting Sugeno fuzzy logic (SFL) technique is used to achieve the dynamic online results of the investigated autonomous power system model for online, off-nominal operational circumstances. Under step perturbations, the time-domain transient investigation in reference to voltage and/or mandate of load for the proposed autonomous power system model is inspected. Additionally, the robustness of the proposed moth flame optimization-based proportional-integral-derivative plus second derivative controller is investigated to test its behavior. An investigation has been provided by varying the model components of the studied autonomous power system model. It may be reported, as per the results obtained from the simulation, that the proposed moth flame optimization-based proportional-integral-derivative plus second derivative controller is an effective control strategy for the autonomous power system. The current research effort indicates that the proposed moth flame optimization algorithm, along with Sugeno fuzzy logic, may be useful for the actual time process of an autonomous power system.

## KEYWORDS

autonomous distributed power system, moth flame optimization (MFO) technique, Sugeno fuzzy logic, real time operation, PIDD 2 controller.



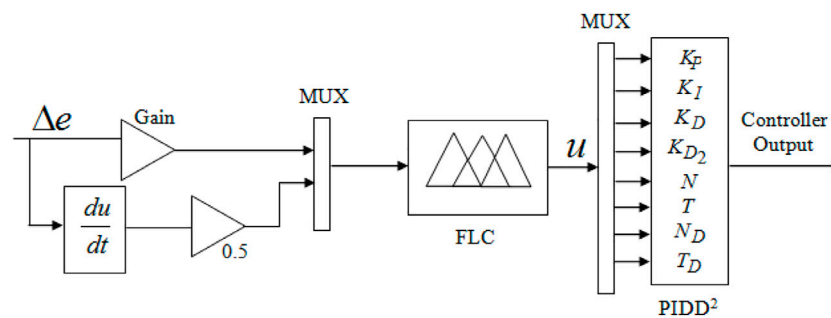


FIGURE 2  
Fuzzy-based PIDD<sup>2</sup> controller.

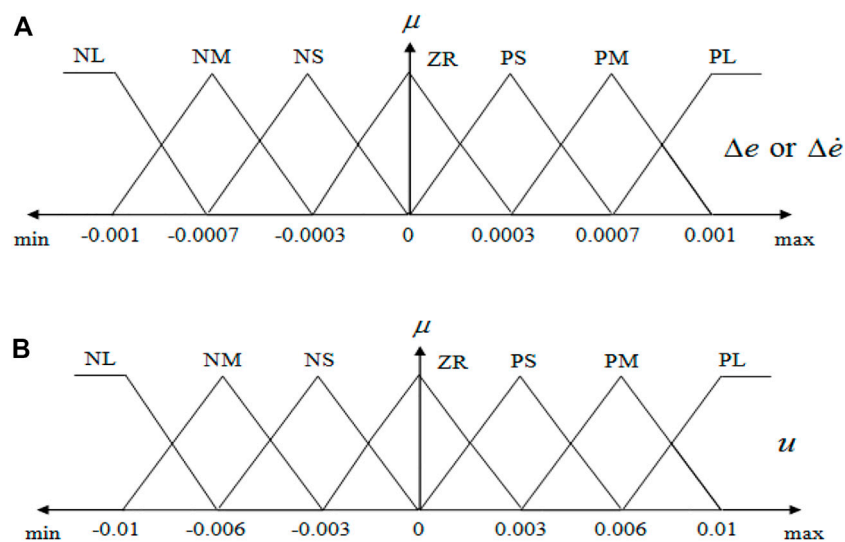


FIGURE 3  
Membership functions of the MFO-SFL-based PIDD<sup>2</sup> controller (A) input ( $\Delta e$  or  $\Delta \dot{e}$ ) and (B) output ( $u$ ).

However, a good balance, particularly between exploration and exploitation, is achieved by hybridization; thus, superior performance is guaranteed when these techniques are combined, as seen in (Al Gizi et al., 2015a; Al Gizi et al., 2015b; Al Gizi, 2019; Ali and Khaniki, 2020; Mokeddem and Mirjalili, 2020; Ozgenc et al., 2020). Hence, this work is focused on modeling and optimizing an autonomous distributed power system using a high-performance proportional-integral-derivative-second order derivative (PIDD<sup>2</sup>) controller improved using the Moth flame optimization (MFO) technique (Mirjalili, 2015) and/or in conjunction with real-time Sugeno fuzzy logic (SFL). MFO was suggested and well deliberated in Mirjalili (2015), and has been used since then in different fields (Shehab et al., 2020), including power systems, as seen in Mohanty (2019), Chatterjee et al. (2020), Chatterjee and Mohammed (2022). Moreover, MFO variants were well studied in literature; the authors of (Nadimi-Shahraki et al., 2021a) propose a migration-based moth-flame optimization (M-MFO) technique that is projected to avoid risk of local optima entrapment. The primary focus of M-MFOs is

on taming the location of unlucky moths by traveling them stochastically in the initial iterations using an arbitrary migration (RM) operator, sustaining the result modification by distinctly storing new skilled results in a controlling archive, and finally manipulating around the locations protected in the controlling archive using a guided migration (GM) operator. Further, an improved moth-flame optimization (I-MFO) algorithm is proposed by the authors of (Nadimi-Shahraki et al., 2021b) to deal with established MFO problems by positioning stuck moths in local optima by defining memory for each moth. The stuck moths have a tendency to eliminate the local optima by using the enhanced wandering around search (AWAS) strategy. Also, (Nadimi-Shahraki et al., 2022) postulated an operational hybridization of the whale optimization algorithm (WOA) and an improved moth-flame optimization algorithm (MFO) named WMFO to resolve the OPF problem. The WOA and the improved MFO work together in the WMFO to effectively determine capable zones and offer high-quality results.

TABLE 1 Control rules of the MFO-SFL-based PID<sup>2</sup> controller.

Input 1→ input 2↖	NL	NM	NS	ZR	PS	PM	PL
NL	PL	PL	PL	PM	PM	PS	ZR
NM	PL	PM	PM	PM	PS	ZR	NS
NS	PL	PM	PS	PS	ZR	NS	NM
ZR	PM	PM	PS	ZR	NS	NM	NM
PS	PM	PS	ZR	NS	NS	NM	NL
PM	PS	ZR	NS	NM	NM	NM	NL
PL	ZR	NS	NM	NM	NL	NL	NL

It should be noted that the introduction of SFL to carry parameter specifications in real-time makes the techniques presented here behave as if they are parameter-free, allowing for easy achievement of good balance. Consequently, the main objectives of this work are:

- a) To tune the offline parameters of the PID<sup>2</sup> controller using the MFO technique for the APS being considered.
- b) To explore the utility and applicability of SFL-based controller tuning in online real-time situations.
- c) To compare the voltage profile response obtained from the proposed technique with that acquired by other studied techniques.
- d) To investigate the model’s performance for a wide range of important system parameters and disturbances for practical implementation.

## 2 Methodology

### 2.1 Power system modeling

A standard distributed energy generator (DEG) consisting of an AVR, speed governor, and controller is depicted in Figure 1; the upper half of the blocks are the mechanical model of the speed

governor with an integrator of gain  $K_{ii}$  and droop  $R$ , while the lower half of the blocks are the electrical model of the AVR system of the studied DEG system with a PID<sup>2</sup> controller. The integral controller eliminates the studied system’s steady-state frequency error, which is modeled by the transfer function given in Eq. 1.

$$G_{Integral} = \frac{K_{ii}}{s} \tag{1}$$

The generator inertia ( $H$ ), damping constant ( $D$ ), and governor controller variables are the key features influencing frequency aberration. On a diesel engine, the inertia constant ( $H$ ) can be documented as the ratio of energy deposited in revolving components of the diesel generator to the rated apparent power, whereas the actuator is normally implemented to refer to the fuel system’s actuator, which controls the volume of diesel fuel inserted into the engine. The transfer function of Eqs 2–4 models the DEG, valve actuator, inertia, and load of the power system under consideration, where  $\tau_D$  and  $\tau_V$  are respectively the time constant of the diesel generator system and valve actuator system while  $H$  and  $D$  are respectively the inertia and damping constants of the DEG under consideration (Banerjee et al., 2012).

$$G_{DEG} = \frac{1}{1 + s\tau_D} \tag{2}$$

$$G_{VA} = \frac{1}{1 + s\tau_V} \tag{3}$$

$$G_{IL} = \frac{1}{2Hs + D} \tag{4}$$

The AVR system comprises four major parts: a generator, an exciter, an amplifier, and a sensor. The sensor detects the voltage at the synchronous generator’s terminal indefinitely. The signal is rectified and smoothed and then directed to the comparator for comparison with a pre-set signal. The comparator’s voltage error is then amplified through the amplifier and directed to the exciter to regulate the windings of an alternator field. Furthermore, the transfer function of an amplifier, exciter, and generator is modeled by a gain and a time constant, as shown in Eqs 5–7, where  $K_A$  and  $\tau_A$ ,  $K_E$  and  $\tau_E$  and  $K_G$  and  $\tau_G$ , respectively, denote the gain and the time constant of the amplifier, modern exciter, and generator systems (Banerjee et al., 2012).

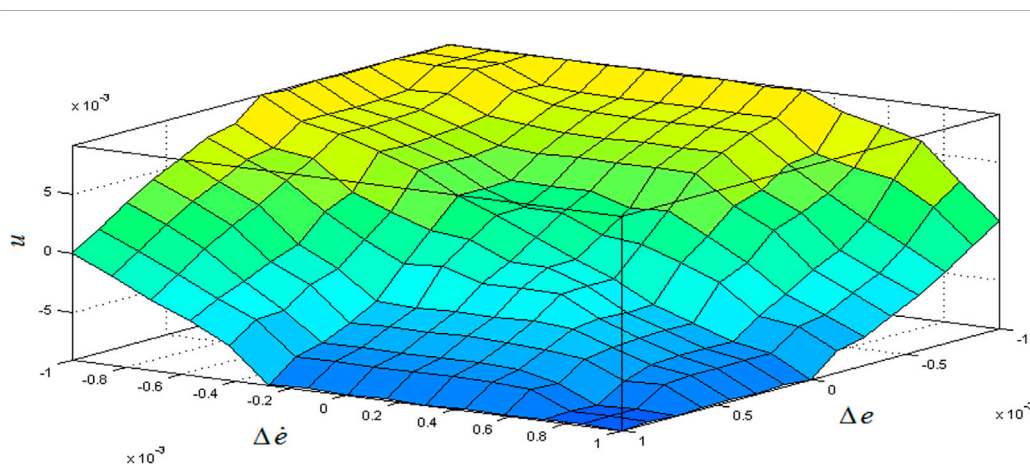


FIGURE 4 Signal surface plot with  $(\Delta e)$  and  $(\Delta \dot{e})$ .

TABLE 2 MFO-optimized controller gains and transient response parameters with varying  $K_G$  and  $\tau_G$ .

$K_G$	$\tau_G$	Controller type	$K_P$	$K_I$	$K_D$	$K_{D_2}$	$N$	$T$ (s)	$N_D$	$T_D$ (s)	$T_S$ (s)	$T_R$ (s)	$M_P$ (p.u.)	FOD
0.7	1.2	<b>MFO_PIDD<sup>2</sup></b>	<b>1.3100</b>	<b>0.8000</b>	<b>0.4700</b>	<b>0.0404</b>	<b>29.000</b>	<b>0.0100</b>	<b>29.000</b>	<b>0.02</b>	<b>2.4781</b>	<b>0.2983</b>	<b>0.0102</b>	<b>0.8082</b>
		MFO_PID	0.1015	0.0900	0.0050	—	—	—	—	—	5.4944	2.2227	0.0102	1.2098
		TLBO_PID	0.1003	0.0855	0.0033	—	—	—	—	—	5.8356	2.3646	0.0101	1.2830
		QOHS_PID	0.1010	0.0802	0.0050	—	—	—	—	—	6.6497	2.5743	0.0101	1.5054
		GA_PID	0.0644	0.0525	0.0050	—	—	—	—	—	10.3079	4.6189	0.0101	2.0989
	1.4	<b>MFO_PIDD<sup>2</sup></b>	<b>1.3100</b>	<b>0.8500</b>	<b>0.4700</b>	<b>0.0420</b>	<b>29.000</b>	<b>0.0200</b>	<b>26.000</b>	<b>0.01</b>	<b>3.2309</b>	<b>0.3067</b>	<b>0.0102</b>	<b>1.0819</b>
		MFO_PID	0.1090	0.0850	0.0049	—	—	—	—	—	5.7893	2.4112	0.0102	1.2491
		TLBO_PID	0.1000	0.0800	0.0040	—	—	—	—	—	6.0849	2.6057	0.0102	1.2861
		QOHS_PID	0.1000	0.0801	0.0050	—	—	—	—	—	6.1007	2.6041	0.0102	1.2925
		GA_PID	0.1000	0.0763	0.0050	—	—	—	—	—	6.5141	2.7605	0.0101	1.3869
0.8	1.2	<b>MFO_PIDD<sup>2</sup></b>	<b>1.3100</b>	<b>0.8000</b>	<b>0.4700</b>	<b>0.0420</b>	<b>30.052</b>	<b>0.0131</b>	<b>29.461</b>	<b>0.01</b>	<b>2.4257</b>	<b>0.2519</b>	<b>0.0102</b>	<b>0.8059</b>
		MFO_PID	0.1080	0.0850	0.0040	—	—	—	—	—	5.1820	1.9656	0.0101	1.8887
		TLBO_PID	0.1000	0.0819	0.0050	—	—	—	—	—	5.3543	2.0757	0.0101	1.2123
		QOHS_PID	0.1000	0.0803	0.0045	—	—	—	—	—	5.4973	2.1237	0.0101	1.2472
		GA_PID	0.0889	0.0644	0.0138	—	—	—	—	—	8.6800	2.9788	0.0101	2.1033
	1.4	<b>MFO_PIDD<sup>2</sup></b>	<b>1.3110</b>	<b>0.8000</b>	<b>0.4700</b>	<b>0.0390</b>	<b>29.0662</b>	<b>0.02</b>	<b>30.5356</b>	<b>0.01</b>	<b>3.3526</b>	<b>0.2889</b>	<b>0.0103</b>	<b>1.1333</b>
		MFO_PID	0.1100	0.0810	0.0050	—	—	—	—	—	5.3694	2.1320	0.0102	1.1971
		TLBO_PID	0.1091	0.0823	0.0038	—	—	—	—	—	6.8901	2.1049	0.0102	1.7665
		QOHS_PID	0.1000	0.0782	0.0039	—	—	—	—	—	7.7254	2.2560	0.0103	2.0182
		GA_PID	0.0945	0.0762	0.0057	—	—	—	—	—	7.8935	2.3334	0.0103	2.0515
0.9	1.0	<b>MFO_PIDD<sup>2</sup></b>	<b>1.3100</b>	<b>0.8000</b>	<b>0.4700</b>	<b>0.0420</b>	<b>29.000</b>	<b>0.0199</b>	<b>31.00</b>	<b>0.010</b>	<b>2.2865</b>	<b>0.1559</b>	<b>0.0100</b>	<b>0.7899</b>
		MFO_PID	0.1000	0.0850	0.0050	—	—	—	—	—	4.6402	1.6332	0.0100	1.1124
		TLBO_PID	0.1020	0.0819	0.0030	—	—	—	—	—	4.9107	1.6937	0.0101	1.1896
		QOHS_PID	0.1013	0.0801	0.0039	—	—	—	—	—	5.4273	1.7314	0.0101	1.3658
		GA_PID	0.0895	0.0813	0.0050	—	—	—	—	—	6.9489	1.7471	0.0101	1.9196
	1.2	<b>MFO_PIDD<sup>2</sup></b>	<b>1.3101</b>	<b>0.8003</b>	<b>0.4710</b>	<b>0.0427</b>	<b>29.8929</b>	<b>0.0162</b>	<b>30.5595</b>	<b>0.010</b>	<b>2.3558</b>	<b>0.2059</b>	<b>0.0102</b>	<b>0.7972</b>
		MFO_PID	0.1090	0.0850	0.0050	—	—	—	—	—	4.5986	1.6758	0.0102	1.0815
		TLBO_PID	0.1079	0.0833	0.0050	—	—	—	—	—	4.6851	1.7112	0.0102	1.1003

(Continued on following page)

TABLE 2 (Continued) MFO-optimized controller gains and transient response parameters with varying  $K_G$  and  $\tau_G$ .

$K_G$	$\tau_G$	Controller type	$K_P$	$K_I$	$K_D$	$K_{D_2}$	$N$	$T$ (s)	$N_D$	$T_D$ (s)	$T_S$ (s)	$T_R$ (s)	$M_p$ (p.u.)	FOD
		QOHS_PID	0.1000	0.0803	0.0049	—	—	—	—	—	4.8375	1.8106	0.0102	1.1198
		GA_PID	0.0998	0.0801	0.0080	—	—	—	—	—	4.8601	1.8043	0.0101	1.1303
1.0	1.0	MFO_PIDD <sup>2</sup>	<b>1.3100</b>	<b>0.8000</b>	<b>0.4700</b>	<b>0.0420</b>	<b>30.0674</b>	<b>0.0122</b>	<b>29.3991</b>	<b>0.010</b>	<b>2.2011</b>	<b>0.1307</b>	<b>0.0101</b>	<b>0.7678</b>
		MFO_PID	0.1090	0.0850	0.0050	—	—	—	—	—	4.1520	1.3648	0.0101	1.0315
		TLBO_PID	0.0948	0.0801	0.0042	—	—	—	—	—	4.3960	1.5288	0.0101	1.0609
		QOHS_PID	0.1001	0.0800	0.0050	—	—	—	—	—	4.4000	1.4934	0.0101	1.0755
		GA_PID	0.0737	0.0684	0.0050	—	—	—	—	—	5.5157	2.2660	0.0103	1.2018
	1.6	MFO_PIDD <sup>2</sup>	<b>1.3100</b>	<b>0.8000</b>	<b>0.4600</b>	<b>0.0320</b>	<b>30.4994</b>	<b>0.0139</b>	<b>29.6355</b>	<b>0.0219</b>	<b>3.3917</b>	<b>0.2381</b>	<b>0.0105</b>	<b>1.1665</b>
		MFO_PID	0.1111	0.0713	0.0040	—	—	—	—	—	6.9519	1.8882	0.0103	1.8689
		TLBO_PID	0.1097	0.0745	0.0043	—	—	—	—	—	7.7121	1.8239	0.0105	2.1723
		QOHS_PID	0.1100	0.0800	0.0047	—	—	—	—	—	8.0307	1.7309	0.0107	2.3239
		GA_PID	0.0999	0.0497	0.0050	—	—	—	—	—	11.2313	4.3695	0.0101	2.4599

$$G_{Amplifier}(s) = \frac{K_A}{1 + s\tau_A} \tag{5}$$

$$G_{Exciter}(s) = \frac{K_E}{1 + s\tau_E} \tag{6}$$

$$G_{Generator}(s) = \frac{K_G}{1 + s\tau_G} \tag{7}$$

### 2.2 Controller modeling

Unlike the conventional PID controller, which has three main components for improving system performance, the PIDD<sup>2</sup> controller employed here has an additional component called second-order derivative gain ( $K_{D_2}$ ) for better performance. As a result, the four parameters, namely,  $K_P$ ,  $K_I$ ,  $K_D$  and a second-order derivative gain ( $K_{D_2}$ ), must be carefully modeled according to the transfer function expressed by Eq. 8.

$$G_{PIDD^2} = K_P + \frac{K_I}{s} + sK_D + s^2K_{D_2} \tag{8}$$

However, two low-pass filters have been employed in the controller design for smoother and better response; these are the first-order low-pass filter for first-order derivatives and a second-order low-pass filter for second-order derivatives in their corresponding tracks, as shown in Eqs 9, 10. Thus, the overall transfer function of the employed PIDD<sup>2</sup> controller can be represented by Eq. 11 where  $N$  and  $N_D$  are the filter coefficients and  $T$  and  $T_D$  are the time constants.

$$G_{First\_Filter}(s) = \frac{s}{s\left(\frac{1}{N}\right)T + 1} \tag{9}$$

$$G_{Second\_Filter}(s) = \frac{s^2}{s^2\left(\frac{1}{N_D}\right)^2T_D^2 + 2s\left(\frac{1}{N_D}\right)T_D + 1} \tag{10}$$

$$G_{PIDD^2}(s) = K_P + \frac{K_I}{s} + sK_D \left( \frac{1}{s\left(\frac{1}{N}\right)T + 1} \right) + s^2K_{D_2} \left( \frac{1}{s^2\left(\frac{1}{N_D}\right)^2T_D^2 + 2s\left(\frac{1}{N_D}\right)T_D + 1} \right) \tag{11}$$

The suitability of the proposed technique is assessed by a flexible time domain performance index known as the “figure of merit” (FOD), represented in Eq. 12; it is implemented using the system’s essential dynamic attributes based on desired specifications and constraints. Consequently, the main target of this optimization task is to minimize FOD, which is directly dependent on the system’s transient response parameters. It should be noted that the  $\beta$  magnitude in Eq. 12 is set to 1.0. Therefore, based on Eq. 12, the design constraints are the controller parameter limits, which are characterized by Eq. 13 as the set of minimum and maximum specified design variables.

$$FOD = (1 - e^{-\beta}) (M_p + E_{SS}) + e^{-\beta} (T_S - T_R) \tag{12}$$

$$\left. \begin{aligned} K_P^{\min} &\leq K_P \leq K_P^{\max} \\ K_I^{\min} &\leq K_I \leq K_I^{\max} \\ K_D^{\min} &\leq K_D \leq K_D^{\max} \\ K_{D_2}^{\min} &\leq K_{D_2} \leq K_{D_2}^{\max} \\ N^{\min} &\leq N \leq N^{\max} \\ T^{\min} &\leq T \leq T^{\max} \\ N_D^{\min} &\leq N_D \leq N_D^{\max} \\ T_D^{\min} &\leq T_D \leq T_D^{\max} \end{aligned} \right\} \tag{13}$$

TABLE 3 Robustness examination for MFO-based PIDD<sup>2</sup> technique.

Model parameters	Change in rate (%)	$M_P$ (p.u.)	$T_S$ (s)	$T_R$ (s)
$\tau_A$	-50%	0.0100	2.4829	0.0918
	-25%	0.0100	2.4238	0.1119
	+25%	0.0104	2.4426	0.1445
	+50%	0.0107	2.6094	0.1586
$\tau_E$	-50%	0.0110	2.8215	0.0534
	-25%	0.0102	2.5513	0.0883
	+25%	0.0103	2.8618	0.1697
	+50%	0.0106	3.1287	0.2077
$\tau_G$	-50%	0.0117	2.9629	0.0491
	-25%	0.0105	2.4752	0.0845
	+25%	0.0102	2.8521	0.1821
	+50%	0.0104	3.3807	0.2360
$\tau_S$	-50%	0.0100	2.4332	0.2079
	-25%	0.0100	2.3733	0.1577
	+25%	0.0104	2.3369	0.1170
	+50%	0.0110	2.4157	0.1087

### 2.3 SFL for online tuning of the PIDD<sup>2</sup> controller

Fuzzy control deviates from conventional control theories to a greater extent because it attempts to model logical reasoning with a vague statement rather than the usual true or false. Sugeno fuzzy logic (SFL) is an efficient fuzzy control technique for a system with fast-changing dynamics like AVR; it can modify the controller’s parameters with precision even online and in a real-time environment. Real-time operations guarantee that balance is achieved between load and generation at all times by making the best use of asynchronous data and control to accommodate changing conditions and delays in communication; fuzzy control is extensively discussed in (Mohagheghi and Harley, 2004). Figure 2 shows the structure of a fuzzy logic-based PIDD<sup>2</sup> controller, the implementation of which necessitates two inputs to yield a control signal ( $u$ ): incremental change in error ( $\Delta e$ ), and derivative of the incremental change in error, ( $\Delta \dot{e}$ ).

The incremental change in error shown in Eq. 14 is the difference between the change in reference voltage and the change in terminal voltage, whereas the derivative of incremental change in error shown in Eq. 15 is the rate of change of incremental change in error, where  $\Delta e_i$  and  $\Delta e_{i-1}$  are the incremental change in error at the time  $i$  and  $(i - 1)$ , respectively, while  $t$  is the sample time in seconds.

$$\Delta e = (\Delta V_{ref} - \Delta V_s) \tag{14}$$

$$\Delta \dot{e} = \frac{\Delta e_i - \Delta e_{i-1}}{t} \tag{15}$$

These two inputs ( $\Delta e$ ;  $\Delta \dot{e}$ ) are split into seven fuzzy classes: positive large (PL), positive medium (PM), positive small (PS), zero (ZR), negative small (NS), negative medium (NM), and negative large (NL). The membership functions of this fuzzy control, which show the degree of membership of the real variable in the corresponding fuzzy variable, are displayed in Figure 3 for the input ( $\Delta e$ ) and ( $\Delta \dot{e}$ ) for the output ( $u$ ). The membership function is made up of two trapezoidal memberships and five triangular membership components, making up seven MFs with two inputs and one output, as depicted in Figure 4 and the corresponding control values specified in Table 1. Therefore, as indicated in Table 1, the output MF is determined by two input MFs in each cell of the control rule. Hence the control rules are realized as follows: if input 1 and input 2 are both true, then output 1 is true. Figure 4 depicts the surface plot profile of the control signal with the deviation of the incremental change in error and the derivative of the incremental change in the error signal.

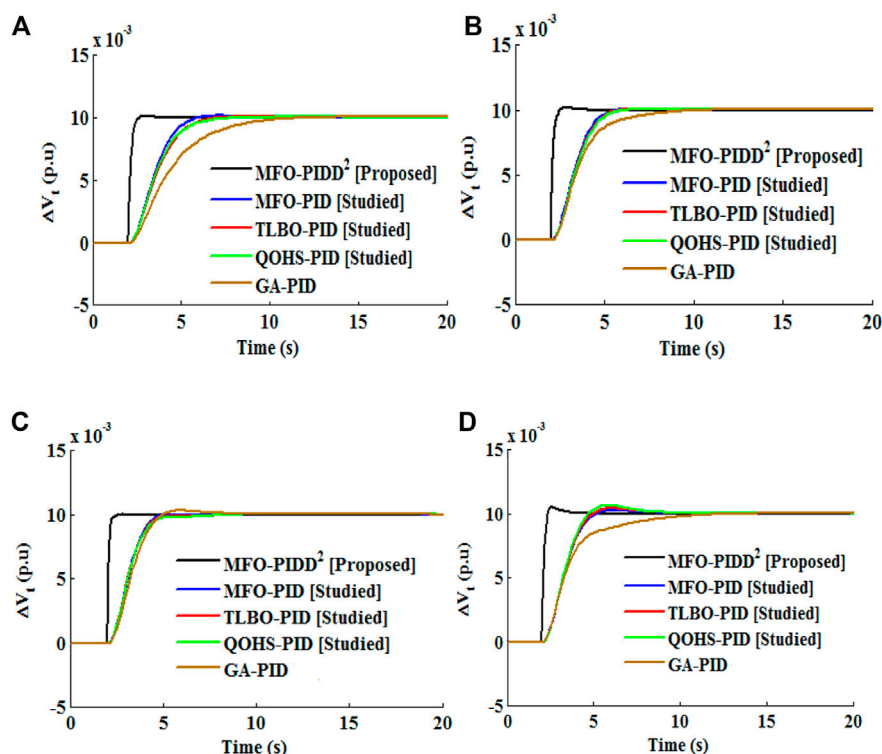
### 3 MFO overview

Moth Flame Optimization (MFO) is inspired by moth night navigation features; fundamentally, MFO mimics the unwanted behavior of moths while navigating in the presence of artificial light and performs its optimization. MFO was proposed and extensively discussed in (Mirjalili, 2015), and it has since been used in a variety of fields (Shehab et al., 2020), including power systems, as seen in (Mohanty, 2019; Chatterjee et al., 2020; Chatterjee and Mohammed, 2022). It is a population-based algorithm that employs two important components: moths and

TABLE 4 SFL-based controller gains, Transient characteristics, and FOD values.

$K_G$	$\tau_G$	Controller type	$K_P$	$K_I$	$K_D$	$K_{D_2}$	$N$	$T$ (s)	$N_D$	$T_D$ (s)	$T_S$ (s)	$T_R$ (s)	$M_P$ (p.u.)	FOD
0.72	1.42	<b>MFO_PIDD<sup>2</sup></b>	<b>1.4883</b>	<b>0.8993</b>	<b>0.5100</b>	<b>0.0415</b>	<b>30.0334</b>	<b>0.01</b>	<b>31.7791</b>	<b>0.0156</b>	<b>3.3483</b>	<b>0.2895</b>	<b>0.0104</b>	<b>1.1316</b>
		MFO_PID	0.1488	0.0878	0.0042	—	—	—	—	—	6.9115	2.0992	0.0101	1.7764
		TLBO_PID	0.1553	0.0901	0.0048	—	—	—	—	—	7.1049	1.9989	0.0101	1.8844
		QOHS_PID	0.1530	0.088	0.0040	—	—	—	—	—	7.3151	2.0742	0.0101	1.9340
		GA_PID	0.0798	0.0525	0.0050	—	—	—	—	—	10.3273	4.4918	0.0101	2.1527
0.87	1.89	<b>MFO_PIDD<sup>2</sup></b>	<b>1.3100</b>	<b>0.6103</b>	<b>0.5033</b>	<b>0.0320</b>	<b>29.1134</b>	<b>0.011</b>	<b>28.7601</b>	<b>0.0127</b>	<b>2.8867</b>	<b>0.3128</b>	<b>0.0102</b>	<b>0.9531</b>
		MFO_PID	0.2101	0.0902	0.0030	—	—	—	—	—	4.3495	1.4952	0.0101	1.0562
		TLBO_PID	0.1889	0.0889	0.0050	—	—	—	—	—	4.4952	1.6037	0.0102	1.0635
		QOHS_PID	0.1811	0.0815	0.0045	—	—	—	—	—	4.8687	1.7436	0.0101	1.1632
		GA_PID	0.0881	0.0516	0.0050	—	—	—	—	—	7.6373	3.4715	0.0102	1.5387
0.95	1.67	<b>MFO_PIDD<sup>2</sup></b>	<b>1.3003</b>	<b>0.7002</b>	<b>0.5055</b>	<b>0.0331</b>	<b>29.0334</b>	<b>0.02</b>	<b>30.9911</b>	<b>0.0145</b>	<b>2.7182</b>	<b>0.2459</b>	<b>0.0103</b>	<b>0.9159</b>
		MFO_PID	0.1600	0.08119	0.0020	—	—	—	—	—	4.4839	1.5766	0.0101	1.0757
		TLBO_PID	0.1553	0.08011	0.0030	—	—	—	—	—	4.5448	1.6153	0.0101	1.0838
		QOHS_PID	0.1430	0.0788	0.0029	—	—	—	—	—	4.6596	1.7114	0.0102	1.0908
		GA_PID	0.0948	0.0555	0.0050	—	—	—	—	—	6.8742	2.7631	0.0101	1.5185
1.01	1.96	<b>MFO_PIDD<sup>2</sup></b>	<b>1.4153</b>	<b>0.8993</b>	<b>0.5100</b>	<b>0.0415</b>	<b>30.0001</b>	<b>0.0213</b>	<b>29.9911</b>	<b>0.0219</b>	<b>3.9157</b>	<b>0.2735</b>	<b>0.0106</b>	<b>1.3463</b>
		MFO_PID	0.1730	0.0799	0.0048	—	—	—	—	—	5.5550	1.5266	0.0103	1.4882
		TLBO_PID	0.1488	0.0788	0.0042	—	—	—	—	—	7.0817	1.6834	0.0105	1.9922
		QOHS_PID	0.1411	0.0766	0.0045	—	—	—	—	—	7.4583	1.7592	0.0105	2.1028
		GA_PID	0.0762	0.0315	0.0050	—	—	—	—	—	13.0286	5.6841	0.0101	2.7077





**FIGURE 5** Comparison of transient characteristics for (A)  $K_G = 0.7$ ;  $\tau_G = 1.2$  s, (B)  $K_G = 0.8$  and  $\tau_G = 1.2$  s, (C)  $K_G = 0.9$ ;  $\tau_G = 1.0$  s and (D)  $K_G = 1.0$  and  $\tau_G = 1.6$  s.

flames. Therefore, it is randomly initiated in space by establishing candidate solutions (moths), then computing their fitness levels and aligning with the flames, which is the best position moths can reach. Afterward, exploitation and exploration are adaptively ensured by a progressive and systematic reduction in the number of flames; thus, for a moth to attain a greater position, it must update its position over and over again according to the position of the flame until convergence is achieved, at which point the process is terminated. The relevant equations required for executing the algorithm are established in (Mirjalili, 2015; Shehab et al., 2020; Chatterjee and Mohammed, 2022), and the general formulation for implementing the MFO algorithm as used in APS is specified in Algorithm 1, (Chatterjee and Mohammed, 2022).

### 3.1 MFO algorithm

The MFO algorithm assumes that the moths are the candidate result and that their location in space is the variable quantity of the problem. As a result, the moths can fold away in 1- $d$ , 2- $d$ , 3- $d$ , or hyper-dimensional space by changing their location vectors. Because this is a population-based algorithm, the set of moths is described in a matrix that is well-defined in Eq. 16 (Mirjalili, 2015)

$$M = \begin{bmatrix} m_{1,1} & m_{1,2} & \dots & \dots & m_{1,d} \\ m_{2,1} & m_{2,2} & \dots & \dots & m_{2,d} \\ \cdot & \cdot & \cdot & \cdot & \cdot \\ \cdot & \cdot & \cdot & \cdot & \cdot \\ m_{n,1} & m_{n,2} & \dots & \dots & m_{n,d} \end{bmatrix} \quad (16)$$

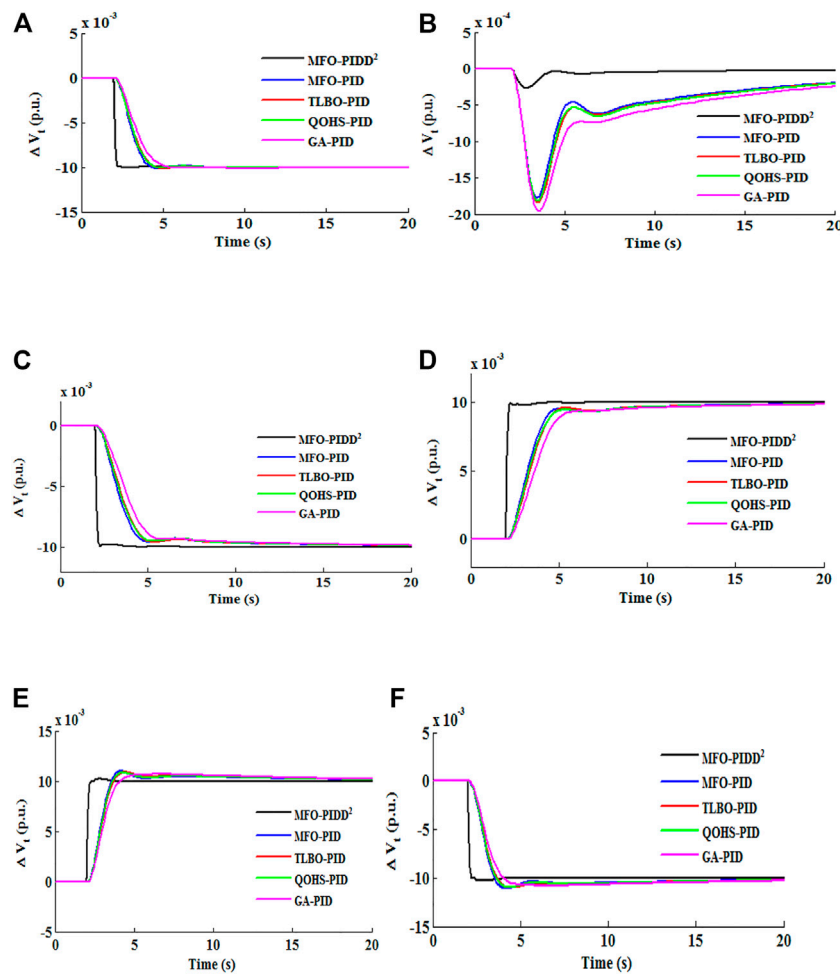
Where  $n$  is the moth numeral data, and  $d$  is the variable numerical data (dimension).

It is assumed that there is an array of the corresponding fitness magnitudes for all moths, as stated in Eq. 17 (Mirjalili, 2015).

$$OM = [OM_1 \quad OM_2 \quad \dots \quad OM_n]^T \quad (17)$$

It should be noted that the fitness magnitude for each moth will be the return magnitude of the fitness (objective) function. The fitness function sends the location vector (1st row in the matrix  $M$ , for example,) of an individual moth, which then conveys the fitness function's output with the corresponding moth as its fitness magnitude ( $OM_1$  in the matrix  $OM$  well-defined in Eq. 17, for instance).

The MFO algorithm has an additional critical component known as flames, for which, similar to the moth matrix, an additional matrix is used as shown in Eq. 18 (Mirjalili, 2015).



**FIGURE 6** Comparison of transient characteristics for distinct disturbances at  $K_G = 1.0$  and  $\tau_G = 1.0$  s. (A)  $\Delta V_{ref} = -0.01, \Delta P_d = 0.00$ , (B)  $\Delta V_{ref} = 0.00, \Delta P_d = -0.01$ , (C)  $\Delta V_{ref} = -0.01, \Delta P_d = 0.01$ , (D)  $\Delta V_{ref} = 0.01, \Delta P_d = -0.01$ , (E)  $\Delta V_{ref} = 0.01, \Delta P_d = 0.01$ , and (F)  $\Delta V_{ref} = -0.01, \Delta P_d = -0.01$ .

$$F = \begin{bmatrix} F_{1,1} & F_{1,2} & \dots & F_{1,d} \\ F_{2,1} & F_{2,2} & \dots & F_{2,d} \\ \cdot & \cdot & \cdot & \cdot \\ \cdot & \cdot & \cdot & \cdot \\ F_{n,1} & F_{n,2} & \dots & F_{n,d} \end{bmatrix} \quad (18)$$

It is clear from Eq. 18 that the size of  $M$  and  $F$  are identical. It is also intended for the flames to have an array for capturing the equivalent fitness magnitudes, as shown in Eq. 19 (Mirjalili, 2015).

$$OF = [OF_1 \quad OF_2 \quad \dots \quad OF_n]^T \quad (19)$$

Individually, the moths and flames (both of which are results) differ because their magnitudes change in each iteration. Moths are actual searching agents that travel from place to place in the search space, with flames being the best location attained so far.

The MFO algorithm is a three-tuple that estimates the global desirable solution for the optimization issues and is expressed in Eq. 20 (Mirjalili, 2015).

$$MFO = (I, P, T) \quad (20)$$

In Eq. 20,  $I$  is a function that explains the fitness magnitudes of an arbitrary population of moths. The analytical model of this function is provided in Eq. 21 (Mirjalili, 2015).

$$I: \phi \rightarrow \{M, OM\} \quad (21)$$

The function  $P$  in Eq. 21 is the main function that transfers the moths in the search space. This function, ultimately, yields a modernized form of the matrix  $M$  after receiving it, as shown in Eq. 22 (Mirjalili, 2015).

$$P: M \rightarrow M \quad (22)$$

If the conclusion condition is fulfilled, then the function  $T$  in Eq. 20 yields true; otherwise, it yields false, as given in Eq. 23 (Mirjalili, 2015).

$$T: M \rightarrow \{true, false\} \quad (23)$$

The overall form of the MFO algorithm is defined in Algorithm 1:

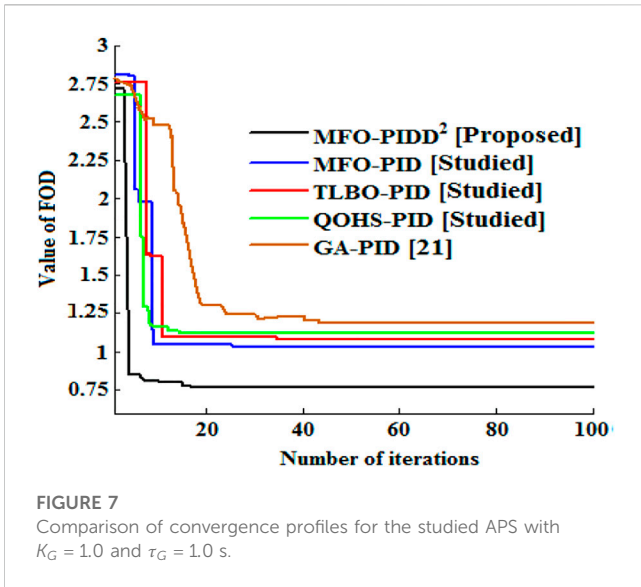


FIGURE 7 Comparison of convergence profiles for the studied APS with  $K_G = 1.0$  and  $\tau_G = 1.0$  s.

```
M=I();
while T(M) is equal to false
    M=P(M);
end
```

Algorithm 1: Overall form of the MFO algorithm (Mirjalili, 2015).

The function  $I$  must compute the objective function magnitudes after producing preliminary results. In this function, the use of any

arbitrary distribution is permitted. The technique shown in Algorithm 2 is employed by default.

```
for i=1:n
    for j=1:d
        M(i,j)=(ub(i)-lb(i))*rand()+lb(i);
    end
end
OM=Fitness Function(M)
```

Algorithm 2: Computation of the objective function (Mirjalili, 2015).

In Algorithm 2, there are two additional arrays named  $ub$  and  $lb$  which outline the upper and lower constraints of the variable quantities, respectively, as shown in Eqs 24, 25 (Mirjalili, 2015).

$$ub = [ub_1, ub_2, ub_3, \dots, ub_{n-1}, ub_n] \quad (24)$$

where  $ub_i$  represents the upper constraint of the  $i^{th}$  variable.

$$lb = [lb_1, lb_2, lb_3, \dots, lb_{n-1}, lb_n] \quad (25)$$

where  $lb_i$  represents the lower constraint of the  $i^{th}$  variable.

The location of individual moths is improved with respect to a flame implementation Eq. 26 (Mirjalili, 2015).

$$M_i = S(M_i, F_j) \quad (26)$$

where  $M_i$  specifies the  $i^{th}$  moth,  $F_j$  specifies the  $j^{th}$  flame, and  $S$  specifies the spiral function.

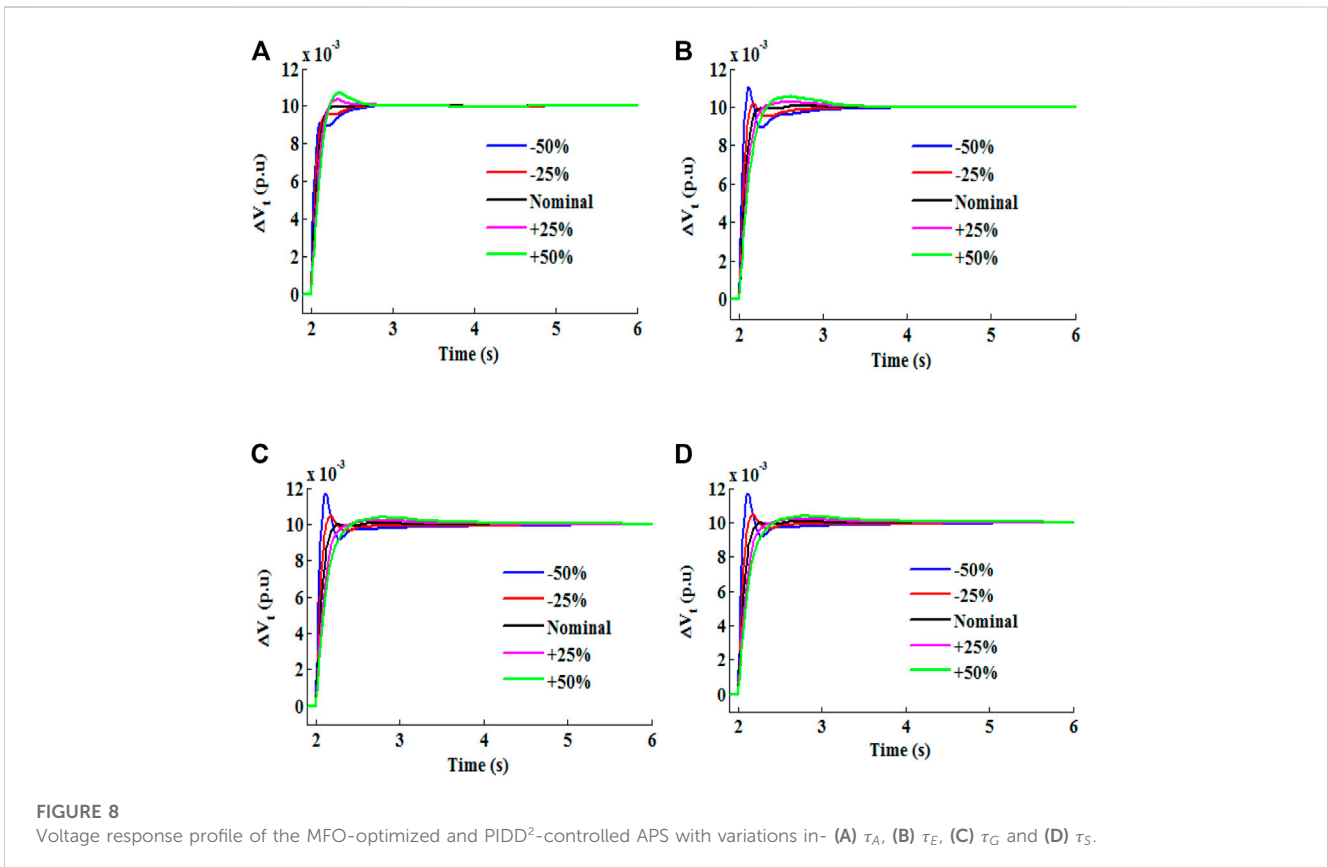


FIGURE 8 Voltage response profile of the MFO-optimized and PIDD<sup>2</sup>-controlled APS with variations in- (A)  $\tau_A$ , (B)  $\tau_E$ , (C)  $\tau_G$  and (D)  $\tau_S$ .

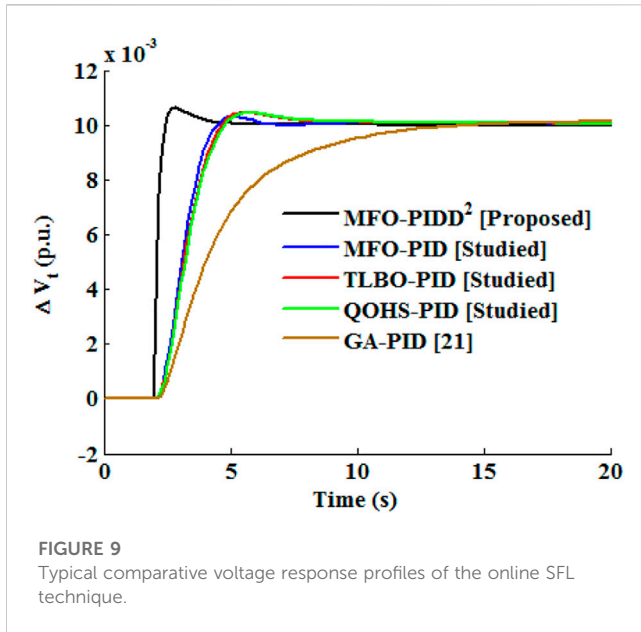


FIGURE 9 Typical comparative voltage response profiles of the online SFL technique.

A logarithmic spiral, as used in the MFO algorithm in (Mirjalili, 2015), is presented in Eq. 27.

$$S(M_i, F_j) = D_i \cdot e^{bt} \cdot \cos(2\pi t) + F_j \quad (27)$$

where  $D_i$  is the gap of the  $i^{\text{th}}$  moth for the  $j^{\text{th}}$  flame,  $b$  is a constant for maintaining the shape of the logarithmic spiral, and  $t$  is an arbitrary number in the range  $[-1, 1]$ .

The variable quantity  $D$  can be designed as per Eq. 28 (Mirjalili, 2015).

$$D_i = |F_j - M_i| \quad (28)$$

where  $M_i$  specifies  $i^{\text{th}}$  moth,  $F_j$  specifies  $j^{\text{th}}$  flame.

The spiral traveling technique of moths is described in Eq. 27. According to the equation, the succeeding location of a moth is described with respect to flame. The spiral technique instructs the moths to inform flames about their locations, making it the main portion of the proposed methodology. The spiral equation allows a moth to hover around a flame. As a result, the search space can be guaranteed for its exploration and exploitation.

The chances of discovering better outcomes can be ensured by taking the best results so far as flames. Hence, the current best results attained are stowed in the matrix  $F$ . Then the moths must modify their locations in relation to the matrix  $F$  throughout the optimization process. It is contrived that  $t$  is an arbitrary number  $[r, 1]$  in order to emphasize additional exploitation, where  $r$  is linearly weakened from  $-1$  to  $-2$  throughout the iteration sequence. In this procedure, the moths become more adept at exploiting their respective flames more effortlessly, according to the number of iterations.

Similar flames aid the moths in varying their locations. The first moth constantly modifies its location with respect to the best flame, while the last moth modifies its position with respect to the worst flame. In the search space, with respect to  $n$  dissimilar positions, the location-modifying procedure of moths may

damage the exploitation of the best result. An adaptive technique is employed to achieve the desired number of flames.

The equation in Eq. 29 can be implemented in this context (Mirjalili, 2015):

$$\text{flame number} = \text{round} \left( N - l \cdot \frac{N-1}{T} \right) \quad (29)$$

where  $l$  signifies the current number of iterations,  $N$  specifies the maximum number of flames, and  $T$  is the maximum number of iterations. Exploitation and exploration of the search space are well-adjusted by the decrease in the number of flames. The stages of the  $P$  function are illustrated in Algorithm 3. The execution of the  $P$  function continues in order to moves the moth in search space until true magnitude of  $T$  function is reached. Then the finest moth is returned as the finest optimized approached magnitude at the end of  $P$ -function.

```

Update flame number using (29)
OM=Fitness Function (M)
If iteration==1
    F=sort (M);
    OF=sort (OM);
else
    F=sort(Mt-1, Mt);
    OF=sort(Mt-1, Mt);
end
for i=1:n
    for j=1:d
        Update r and t
        Calculate D using (28) with respect to the
        corresponding moth
        Update M(i, j) using (26) and (27) with respect to
        the corresponding moth
    end
end

```

Algorithm 3: Overall stages of function  $P$  (Mirjalili, 2015).

## 4 Results and discussion

An autonomous fuel cell distributed power system depicted in Figure 1, with system parameters presented in Appendix A (Banerjee et al., 2012), is considered here for AVR systematic investigation, along with an exceptional proportional-integral-derivative-second order derivative (PIDD<sup>2</sup>) controller; the controller was tuned both offline and online by MFO and SFL-MFO techniques under varying generator gain and time constants. Also evaluated is system performance (specifically voltage response) under various disturbances and uncertainties. As a result, the important findings of this research are discussed below, with the results of particular interest highlighted in the respective tables. All simulations were carried out in MATLAB/SIMULINK (version 7.10) on a 2.77 GHz, Intel Core™ i7 computer, with the maximum number of iterations and population size set to 100 and 50 for each algorithm.

## 4.1 Comparison of transient response characteristics

In this section, the transient characteristics of the system are investigated and compared under two conditions: distinct gain and time constant of the generator ( $K_G$ ;  $\tau_G$ ) and distinct form of step disturbances ( $\Delta V_{ref}$ ;  $\Delta P_d$ ).

### 4.2 First condition: Distinct $K_G$ and $\tau_G$

In the first case, the transient characteristics of the system with varying gain and generator time constant values for the various techniques considered (proposed and studied) are examined and then compared for the same set of  $K_G$ ;  $\tau_G$ . The results highlighted in Table 2 show the optimized gain and transient parameters obtained by each technique, with an emphasis on the MFO-based PIDD<sup>2</sup> technique (outlined in bold text). The table shows that the MFO-based PIDD<sup>2</sup> controller offers the best values for settling time ( $T_s$ ), rise time ( $T_R$ ), and peak overshoot ( $M_p$ ), resulting in a lower value of FOD being constantly incurred by the proposed MFO technique. This performance is evident from the better voltage response profile obtained by the proposed technique in Figure 5 in comparison to other controllers for the different sets of  $K_G$  and  $\tau_G$ .

### 4.3 Second condition: Distinct $\Delta V_{ref}$ and $\Delta P_d$

In this condition, the transient characteristics of the system for the different techniques considered (proposed and studied) are examined and compared for the distinct values of reference voltage and load perturbation  $\Delta V_{ref}$  and  $\Delta P_d$ . Therefore, either zero, negative or positive forms of  $\Delta V_{ref}$  and  $\Delta P_d$  are applied simultaneously to the system for the apparent value of  $K_G = 1.0$  and  $\tau_G = 1.0s$ . The results, shown in Figure 6, demonstrate how effectively the disturbances are handled by the system, as the system itself was able to adjust, withstand, and fine-tune all the deviations separately and concurrently. Furthermore, Figure 6 clearly illustrates the excellent performance of the proposed technique in suppressing oscillations and recovering faster than the other methodologies studied in this work for the same sets of disturbances. To some extent, these have certainly validated the flexibility of the proposed technique.

### 4.4 Convergence

The comparative convergence profile, which was established by plotting the minimum FOD value against the number of iterations (as shown in Figure 7), demonstrates the high performance of the proposed technique; indeed, it converges faster and offers a lower FOD value as compared to the other techniques considered in this work.

## 4.5 Robustness analysis

Due to parametric uncertainties, models are an imperfect representation of reality. Consequently, a controller that is perfectly tuned to a model may reduce system performance or stability. To avoid this problem, the MFO-based PIDD<sup>2</sup> technique edge is checked using robustness analysis by systematically subjecting the model to broad variations of some essential parameters, such as  $\tau_A, \tau_E, \tau_G$  and  $\tau_S$  in the range of  $\pm 50\%$  and the step of  $\pm 25\%$ . However, as highlighted in Figure 8 and supported by the transient characteristics presented in Table 3, a  $\pm 50\%$  deviation from the nominal specification of the model did not result in the system exceeding requirements or failing to meet intent; this undoubtedly indicates the technique's robustness to stochasticity.

### 4.6 SFL real-time response

Having obtained outstanding results with the MFO-based PIDD<sup>2</sup> technique offline, SFL control is introduced into the system for parameter specification online and in real-time, yielding the online optimum controller gains, transient characteristics, and FOD values at a different set of  $K_G$  and  $\tau_G$  for the different technique (proposed and studied) as shown in Table 4. The optimum controller gains in real-time operation for distinct values of  $K_G$ ;  $\tau_G$  are determined *via* the fuzzy rule and the Sugeno inference system. However, for each set of  $K_G$ ;  $\tau_G$ , the proposed techniques outlined in the bold text offer the minimum value of settling time ( $T_s$ ), rise time ( $T_R$ ), and peak overshoot ( $M_p$ ), implying that the proposed technique always incurs a lower FOD value, even online after load perturbation. This outstanding performance can be precisely grasped graphically by comparing the typical voltage response profile obtained by the proposed technique to the different techniques studied (as seen in Figure 9).

## 5 Conclusion

This work used an autonomous distributed power model to conduct a comprehensive investigation. First, the model was investigated offline using the MFO-PIDD<sup>2</sup> technique, which gave an excellent transient response that died out and settled faster than other techniques considered. Furthermore, the comparative convergence and robustness analyses clearly demonstrated the high performance and flexibility of the technique. Also, the study was expanded to operate in online mode by synchronizing real-time SFL with the MFO-PIDD<sup>2</sup>, which gave improved control that outperformed other online techniques used in this study. Lastly, the techniques presented here demonstrate improved resilience with better control and coordination than previous techniques offered in the recent literature on the model; thus, the technique can be used as an effective instrument for enhancing systems with identical or similar specifications.

## Data availability statement

The raw data supporting the conclusion of this article will be made available by the authors, without undue reservation.

## Author contributions

All authors listed have made a substantial, direct, and intellectual contribution to the work and approved it for publication.

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## Conflict of interest

The authors declare that the research was conducted in the absence of any commercial or financial relationships that could be construed as a potential conflict of interest.

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## Appendix A

$K_A = 10.0$ ,  $\tau_A = 0.1$  s,  $K_E = 1.0$ ,  $\tau_E = 0.4$  s,  $K_S = 1.0$ ,  $\tau_S = 0.05$  s,  
 $\tau_D = 2.31$  s,  $\tau_V = 0.82$  s,  $K_1 = 1$ ,  $K_2 = 0.2$ ,  $K_3 = 1.5$ ,  $H = 0.1$ ,  $D = 2.22$ ,

$R = 0.074$  and  $K_{ii} = 0.1000$ . The values of  $K_G$  and  $\tau_G$  are load dependent.



## Glossary

The following abbreviations are used in this document:

$D$	DEG damping constant	$T$	Time constant of the PIDD <sup>2</sup> controller filter, s
$E_{SS}$	Steady-state error, p. u.	$T_D$	PIDD <sup>2</sup> controller time constant, s
$H$	DEG inertia constant	$T_R$	Rise time, s
$K_A$	Constant gain of the amplifier	$T_S$	Settling time, s
$K_D$	Derivative gain of the controller	$\beta$	Constant parameter of the figure of demerit
$K_{D_2}$	Second-order derivative gain of the controller	$\Delta e$	Incremental change in terminal voltage error, p. u.
$K_E$	Constant gain of the exciter	$\Delta e_i$	Incremental change in terminal voltage error at time $i$ , p. u.
$K_G$	Constant gain of the generator	$\Delta e_{i-1}$	Incremental change in terminal voltage error at time $(i - 1)$ , p. u.
$K_{ii}$	Constant gain of the integral speed governor controller	$\Delta \dot{e}$	Derivative of incremental change in error in terminal voltage, p. u.
$K_I$	Integral controller gain	$\Delta V_e(s)$	Error voltage, p. u.
$K_P$	Proportional controller gain	$\Delta V_{ref}(s)$	Incremental change in reference voltage, p. u.
$K_S$	Constant sensor gain	$\Delta V_s(s)$	Feedback voltage, p. u.
$M_p$	Maximum overshoot, p. u.	$\Delta V_t(s)$	Incremental change in terminal voltage, p. u.
$N$	Filter co-efficient of the PIDD <sup>2</sup> controller filter	$\tau_A$	Amplifier time constant, s
$N_D$	Second-order filter co-efficient of the PIDD <sup>2</sup> controller's filter	$\tau_D$	DEG time constant, s
$R$	Constant droop of the speed governor	$\tau_E$	Exciter time constant, s
$t$	Sample time, s	$\tau_G$	Generator time constant, s
		$\tau_S$	Sensor time constant, s
		$\tau_V$	Time constant of the valve actuator, s.