

Precision Flux Control of Linear Reluctance Actuator Using the Integral Sliding Mode Method

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The performance of the actuator is becoming increasingly important in the ultra-precision stage. However, the traditional Lorentz motors with given mechanical parameters cannot provide enough force for the next-generation motion stage in the semiconductor industry since they achieve a physical limit of power factor. To tackle this problem, this study develops a novel-driven approach and its control strategy for high-dynamic stages. Explicitly, the proposed method utilizes a linear reluctance motor as an actuator, which could promote the continuous thrust significantly. A heuristic optimization-based Bouc–Wen model is established to describe the nonlinear behavior of the novel actuator. Also, a flux control algorithm based on the integral sliding mode is derived and adjusted for precision thrust generation. Comparative simulations on a specific linear reluctance motor confirm the effectiveness and superiority of the proposed method and show that it has the ability to conquer the force nonlinearity of the novel actuator.

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Miao Q, Liu Y and Tan JB (2022) Precision Flux Control of Linear Reluctance Actuator Using the Integral Sliding Mode Method. Front. Energy Res. 10:949782. doi: 10.3389/fenrg.2022.949782 Keywords: reluctance actuator, Bouc-Wen model, flux control, integral sliding mode control, TLBO (teaching-learning-based optimization)

1 INTRODUCTION

To increase the throughput, the precision motion stage in lithography machine has to operate at high speed and high acceleration, which requires a lower moving mass or larger driven force (Boeren et al., 2020; Kuang et al., 2020; Huang et al., 2021; Voorhoeve et al., 2021; Kuang et al., 2022). Voice coil motor is widely used in nano-accuracy motion stages because of its advantages in high force linearity and low stiffness as contactless direct actuators (Song et al., 2018; Li L. et al., 2019; Liu et al., 2020; Song et al., 2022). However, this kind of actuator is reaching physical limits considering the maximum force to the equivalent load ratio and the corresponding copper loss ratio. Thus, it is not the best choice for driving the next-generation nano-positioning stage. For example, the reticle stage of a state-of-the-art photolithography scanner must accelerate at tens of g's. Thus, actuators that deliver high force with high efficiency have to be developed.

Electromagnetic technology has been widely concerned because of its advantages of high output density (Xu et al., 2022a; Xu et al., 2022b), a new type of reluctance motor is proposed. Different from the Lorentz motor, the force of the reluctance motor is not linearly proportional but a square of the excitation current. So it could bring about larger force to current ratio and achieve significantly higher power densities and lower dissipations (Cigarini et al., 2020; Ito et al., 2020; Moya-Lasheras et al., 2020; Burgstaller et al., 2021; Moya-Lasheras et al., 2021; Al Saaideh et al., 2022). As a promising new candidate, the reluctance motor also brings various intractable issues. From the control point of view, it includes a variety of nonlinear problems, including the output force proportional to the square of the current, inversely proportional to the stroke of the motor, and the hysteresis effect with

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the multi-value rate. These strong nonlinearities make their control difficult and the force output of reluctance motors cannot be accurately predicted, thus limiting their application in nano motion stages. Among them, the force hysteresis caused by soft ferromagnetic materials is the most notable nonlinearity that has to be considered in precision applications (Cigarini et al., 2019). The idea of this paper is to deal with the multi-loop raterelated hysteresis nonlinearity of the reluctance motor through a smart method, that can achieve the purpose of high-precision control of the flux. This kind of nonlinearity is rarely discussed in permanent magnetic motors, but it exists in piezoelectric actuators widely and attracts lots of attention (Adriaens et al., 2000; Chen et al., 2017; Nie et al., 2022; Zhang et al., 2022). To describe the characteristics of hysteresis, various model structures are proposed. The most representative methods include physical modeling, mathematical operator modeling, and intelligent modeling (Li M. et al., 2019; Al Janaideh et al., 2021; Zhang et al., 2021; Cheng et al., 2022; Flores et al., 2022; Otieno et al., 2022). Specifically, the Bouc-Wen model sorted by the physical method is the most popular one since its parameters have definite physical meaning and are easily obtained (Ismail et al., 2009; Hassani et al., 2014). Once the mathematical expression of the hysteresis is obtained, the corresponding compensation algorithm can be developed. Among the existing methods, one of the most commonly used approaches is the inverse compensation. Qin proposed an adaptive hysteresis compensation method to solve the nonlinearity of PEAs at a high frequency for inexperienced users (Rakotondrabe, 2011). Micky presented an inverse multiplicative structure for the Bouc-Wen based model, reducing the computation load of the preexisting methods (Qin et al., 2022). Didace adopted the inverse Bouc-Wen model as feedforward controller, and the proposed compensation strategy was verified in a 3-DOF piezotube scanner (Habineza et al., 2015). Although the inverse compensation scheme is useful to reduce nonlinearity, a common drawback of this method is that a precise inverse model has to be constructed to counteract the effect of hysteresis. However, hysteresis usually contains more than one



loop (Liu et al., 2016). Thus, the corresponding inverse model with high accuracy is hard to establish.

Different from the inverse compensation method, this paper treats hysteresis as a model uncertainty or a kind of internal disturbance, and develops a robust controller to obtain a precise force output from the reluctance motor. As a popular variable structure control algorithm, the sliding mode control (SMC) makes the system states move towards a predesigned sliding surface and then maintains their movement on the surface (Gu et al., 2015). Once the states reach the sliding surface, the system dynamics is determined by the sliding mode and has a certain robustness to the external disturbance and model variations. Since SMC is easy to implement in embedded systems, this paper will develop an SMC-based control strategy.

The rest of the paper is organized as follows: a novel precision motion stage driven by the linear reluctance motor is described in **section 2**, and the force performance is determined. Then, a Bouc–Wen model is established by using the simulation data and the heuristic algorithm to express the hysteresis relationship between magnetic flux B and current I in **section 3**. Based on



TABLE 1 Set parameters of the TLBO.					
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Parameter	Value	Meaning
Gm	100	Iteration time
NP	50	Population quantity
D	6	Optimization dimension
Ge	0.001	Stop criterion

the established model, an integral sliding mode controller is designed in section 4. Afterward, simulation studies were carried out in section 5, where PI with the inverse hysteresis compensation composite controller was used for comparison. Finally, section 6 concludes this research.

2 WORKING PRINCIPLE AND REQUIREMENT OF LINEAR RELUCTANCE MOTOR IN THE NANO-POSITIONING STAGE

The nano-positioning stage in the lithography machine usually adopts a long-stroke and short-stroke combined structure. The positioning accuracy of the stage is ensured by the short-stroke module, while the long-stroke module provides a large motion range. During the scanning process, the short stroke and long stroke should move coordinately. So, the short-stroke stage also requires high acceleration as well as high accuracy. As mentioned in **section 1**, the linear reluctance motor is a novel candidate actuator in the IC manufacturing field. A testbed driven by a reluctance motor is designed and shown in **Figure 1**.

TABLE 2 | Identified parameters of the classical Bouc–Wen model.

Parameter	Value	Unit
a ₀	-1318.4	[-]
a ₁	1650.6	[-]
a ₂	-635.8727	[-]
α	0.9195	[-]
β	0.5174	[-]
γ	0.2448	[-]

As seen from **Figure 1**, the mover I-beam installed on the motion stage is installed opposite to the stator part of the reluctance actuator. In practice, the two reluctance actuators are energized in turn, and an attractive force is generated between the stator E-core and the mover I-beam, which provides the thrust required for high acceleration. According to the magnetic simulation by COMSOL, the attractive reluctance force between the stator and the I-beam is determined by the magnetic field distribution in the air gap of the linear reluctance motor. As expected, most of the magnetic field distributes around the E-core. If the leakage flux is ignored, the reluctance force F exerted on the mover is described by **the** following equations:

$$B = \frac{\mu_0 NI}{2q},\tag{1}$$

$$F = \frac{B^2 A}{2\mu_0},\tag{2}$$

$$F = \frac{\mu_0 A N^2 I^2}{8g^2},$$
 (3)



where $\mu 0$ denotes the permeability of air, *N* denotes the number of coil turns at the center of the E-core, *A* denotes the area, and g denotes the air gap. It is a lumped model which disregards flux leakage, fringing, and saturation.

The linear reluctance motor is driven by the power amplifier. The current command could be derived from **Eq. 3** as

$$I = \sqrt{\frac{8g^2 \cdot F}{\mu_0 A N^2}}.$$
 (4)

Because the features printed on the IC can be as small as tens of nanometers, the position accuracy of the reticle and wafer during exposure must be accurate to a few nanometers or better. The servo bandwidth is limited by stage dynamics, so the majority of the position accuracy comes from the force feedforward accuracy, which must be better than 99.9% accurate for a typical trajectory (Ian, 2015). Therefore, the fluctuation range of the force must be within the range of 0.1%. Further, the force F is proportional to the square of the magnetic flux B, as shown in **Eq. 3**. So the fluctuation range of the magnetic flux is defined as 0.316%. Therefore, when the reluctance actuator is controlled by magnetic flux feedback, the error is limited to 0.316%, which can meet the requirements.

3 HYSTERESIS MODEL

3.1 Bouc–Wen Model

The Bouc–Wen model was proposed in 1976 and is widely used for describing hysteresis nonlinearity because of its advantages of simple structure and few parameters to be identified. At present, the model is expressed as follows:

$$\begin{cases} x(t) = a_0 x(t) + a_1 I(t) + a_2 h(t) \\ h(t) = \alpha I(t) - \beta |I(t)| h(t) - \gamma I(t)| h(t)|, \\ B(t) = x(t) \end{cases}$$
(5)

TABLE 3 Parameters of ISMC.			
Controller	Parameter	Value	
ISMC	λ	3000	
	ks	1.2	
	Δ	0.1	
	ε	0.2	

where $\mathbf{x}(t)$ is the state variable of the system and $\mathbf{B}(t)$ is the output flux, α, β, γ and a_0, a_1, a_2 are the model parameters. I(t) represents the input current.

3.2 Parameter Identification of Hysteresis Loop

Due to the nonlinearity, it is still a challenge to determine the parameters in the Bouc–Wen model. As an intelligent optimization algorithm, the teaching-learning-based optimization (TLBO) is a preferred method. According to **Eq. 5**, the parameters to be identified are defined by the vector $V = \{a_0, a_1, a_2, \alpha, \beta, \gamma\}$. For the sake of brevity, **Eq. 5** is rewritten in discrete form:

$$B_{BW}(k+1) = F(k, B_{BW}(k), I(k), V),$$
(6)

where $B_{BW}(k) \in \mathbb{R}^q$ stands for the Bouc–Wen model output. The cost function for model identification is the mean square error (MSE), as shown in **Eq. 7**.

$$J_{MSE} = \frac{1}{N} \sum_{i=1}^{N} \left(B_{act}^{i} \left(k \right) - B_{BW}^{i} \left(k \right) \right)^{2}, \tag{7}$$

where N is the total number of samples, $B^i_{BW}(k)$ and $B^i_{act}(k)$ are the Bouc–Wen model output and simulation result, respectively. The model parameters are searched by the TLBO to make the simulated response track the measured data. The identification principle of the hysteresis parameters is shown in the **Figure 2**.



TABLE 4 Parameters of contrastive controller.			
Parameter	Value		
kp	0.8		
	Parameter kp ki		

The process of model construction is carried out offline as follows:

- Data collection: The simulation data including output flux and input current of reluctance motors are obtained by multiphysics simulation software COMSOL.
- (2) Model implementation: The Bouc–Wen model is constructed in the Maltab/Simulink, as shown in Figure 3.
- (3) Parameter identification: The simulation data are loaded, and then the model parameters are optimized by the TLBO.
- (a) The dynamic model for the linear reluctance motor.
- (b) Bouc–Wen model implemented with the Maltab/Simulink block.

4 CONTROLLER DESIGN OF THE RELUCTANCE MOTOR

In this article, we adopt the variable structure control method to solve the nonlinear problem in the reluctance motor caused by hysteresis. Without considering external disturbances, it is assumed that the established Bouc–Wen model can accurately describe the hysteresis characteristics of the reluctance motor. To reduce the adjusting effort of PID, a control algorithm based on the integral sliding mode has been developed.

In the design of the sliding mode controller, the dynamic performance of the system is determined by the sliding mode surface. The sliding mode surface with integral term is widely used since it can effectively eliminate the steady-state error and improve the robustness of the controller. Therefore, an integral-derivative type of sliding surface is defined as follows:

$$s(t) = e(t) + \lambda \int_0^t e(\tau) d\tau,$$
(8)

$$e(t) = x_d(t) - x(t),$$
 (9)

where e(t) stands for the flux error between the desired flux trajectories $x_d(t)$ and actual flux trajectories x(t). $\lambda > 0$ are control gains.

Considering the system described by Eq. 5 and the sliding mode expressed by Eq. 8, if the controller $\lim_{s \in (t) \to 0} s(t) \le 0$ with Eq. 10 is employed, then the flux tracking error Eq. 9 converges to zero asymptotically.

$$I(t) = \frac{1}{a_1} \Big[x_d(t) - a_0 x(t) - a_2 h(t) + \lambda e(t) + k_s |s(t)|^{\varepsilon} \operatorname{sgn}(s(t)) \Big],$$
(10)

where sgn (·) is the symbolic function, $k_s > 0$ represents the rate at which the controller state approaches the sliding surface. ε is an arbitrary positive constant and $0 < \varepsilon < 1$. Furthermore, the robustness and anti-disturbance performance of the sliding mode controller are determined. When $t \to \infty$ the synovial control rate can ensure the stability of the control system and $e(t) \to 0$.

A Lyapunov function is adopted to verify the stability of the closed-loop control system, which is shown as

$$V(t) = \frac{1}{2}s^{2}(t).$$
 (11)

Then, taking the time derivative of the sliding function **Eqs. 8**, **11**:

$$s(t) = e(t) + \lambda e(t), \qquad (12)$$

$$V(t) = \mathbf{s}(t) \cdot \mathbf{s}(t) . \tag{13}$$



Substituting Eq. 12 into Eq. 13 and taking consider of the control input Eq. 5, it leads to

$$V(t) = s(t) \bullet \left(e(t) + \lambda e(t)\right),$$

= $s(t) \bullet \left(x_d(t) - x(t) + \lambda e(t)\right),$ (14)
= $s(t) \bullet \left(x_d(t) - (a_0 x(t) + a_1 u(t) + a_2 h(t)) + \lambda e(t)\right).$

Finally, taking the designed sliding mode control rate Eq. 10 into Eq. 14,

In order to reduce the jitter of the sliding mode control system during the control process, the saturation function sat () was adopted to replace the sign function sgn ().

$$\operatorname{sat}(s(t)) = \begin{cases} \operatorname{sgn}((s(t)) & \operatorname{if} |s(t)| > \Delta \\ s(t)/\Delta & \operatorname{if} |s(t)| \le \Delta \end{cases},$$
(16)

where Δ is positive constant and stand for boundary layer thickness for saturation function.

5 SIMULATION

To verify the feasibility of substituting voice coil motors with reluctance motors, a testbed is designed according to the desired performance. A prototype of a reluctance motor with specified



parameters is designed and tested in COMSOL, and the exported data is used to construct the motor model.

obtained by TLBO. The parameters needed in the optimization is set as Table 1:

To establish the model of the reluctance motor, the Bouc–Wen structure is selected and the corresponding coefficients are

The identification results are listed in **Table 2**. Comparing the actual data and the fitting data using the estimated Bouc–Wen

TABLE 5 | Control performance of different controllers.

Controller	Mean error (T)	Maximum error (T)	Standard deviation (T)
Pure PI control	5.0823e-4	0.0056	0.0028
Combination of PI and hysteresis compensation	1.1222e-5	0.0029	1.1381e-4
ISMC	7.8513e-8	0.0027	4.1896e-5

model, the curves are depicted in **Figure 4**. Due to the model error, the maximal mismatching deviation is 0.005333.

In the simulation, it adopts the flux trajectory as the reference input directly, expressed as $B_d = A \sin(wt) = 1.83 * \sin(7.8125 * t)$, where A denotes the maximum value of input and is calculated by mass and maximum acceleration value and **Eq. 2**, w stands for frequency and it is calculated by maximum acceleration value and maximum jerk value. The relationship between the maximum acceleration and the maximum jerk is $J_{max} = w*a_{max}$. Considering that the maximum acceleration is 12.8 g, the maximum jerk is $1000m/s^3$ Then, w = 7.8125rad/s. Based on the obtained Bouc–Wen model, an integral sliding mode controller is designed to reduce the influence of the nonlinearity, including the hysteresis and uncertainty, then the robustness of the system is enhanced. After constant attempts, the control parameters are determined by learning algorithms (Xu et al., 2021). The results are shown in **Table 3**.

For comparison, the combined strategy of PI controller and inverse model compensation is selected. The schematic diagram is shown in **Figure 5** and the relevant parameters are shown in **Table 4**.

With the proposed ISMC, the output of the simulated reluctance motor is shown in **Figure 6A**. As seen from **Figure 6B**, the width of the hysteresis loop has been reduced, which means that the hysteresis effect has been compensated significantly. In addition, **Figure 6C** depicted the sliding variable, implying that the sliding surface is maintained within the boundary layer Δ (i.e., 0.1). Moreover, the control signal is described in **Figure 6D**. Obviously, there is no chattering in the designed controller.

The flux tracking error results between the input Bd and output B with the two controllers are compared in Figure 7. As seen from Figures 7B,D,F the performance of the proposed ISMC is superior to that of both conventional PI controllers and combined controllers with PI and feedforward hysteresis compensation in terms of tracking errors. When using a PI controller, the flux tracking error ranges from -0.0041T to 0.0056 T. Therefore, the required magnetic flux requirements can not be met. When using PI and hysteresis inverse model compensation control, the flux tracking error ranges from -0.000395T to 0.0029 T. Although inverse hysteresis compensation is employed to counteract the nonlinearity, whereas, when using the proposed ISMC, flux tracking error substantially reduces from 6.287e-06 T- to 0.0027 T. Compared with the combined controller, the error is not only small in amplitude but also has a more stable change trend. Therefore, the integral sliding mode controller has better robustness.

In order to explore the influence of different controllers on the flux tracking performance, the error data obtained by the simulation is analyzed, and the average value, maximum value, and standard deviation of the magnetic flux error under different controllers are obtained. The analysis results are shown in **Table 5**.

When using different feedback controllers, the tracking error of the magnetic flux is shown in **Figure 7** and described in **Table 5**. When using the integral sliding mode controller, compared with the PI controller and the inverse model compensation joint control, the mean, maximum, and standard deviation of the flux tracking errors are reduced by 99.3%, 6.9%, and 68.19%, respectively. Hence, compared with the PI controller and the PI and hysteresis inverse model compensation controller it can be seen that the flux with Integral Sliding Mode Control has better tracking performance and robustness.

6 CONCLUSION AND DISCUSSION

In this paper, the reluctance actuator driven nano-positioning stage was designed and the precision motion control scheme with ISMC was applied. Based on the Bouc–Wen model identified by experimental data and TLBO, the ISMC was designed and the hysteresis characteristics of the reluctance actuator is treated as the internal disturbance. The simulation results show that the proposed method is effective in overcoming the hysteresis without accurate parameters and it is promising in highprecision control applications.

This paper also provides a technical solution in other precision machine tools and related equipment with hysteresis.

DATA AVAILABILITY STATEMENT

The original contributions presented in the study are included in the article/Supplementary Material; further inquiries can be directed to the corresponding author.

AUTHOR CONTRIBUTIONS

YL conceived and designed the simulations; QM performed the simulations; QM analyzed the data; and JBT contributed the investigated stage.

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