



# Numerical Simulation of Flame Combustion Characteristics of Tail Gas From the Copper Slag Depleted Reduction

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The tail gas with relatively high calorific value, generated from the depleted reduction of copper slag, can be used as fuel for gas turbines. In this study, the flame speed calculation module in PREMIX of CHEMKIN was used to simulate the combustion characteristics of the tail gas after sulfide removal, the effects of equivalence ratio, gas preheating temperature, and the CO<sub>2</sub> concentration on the flame propagation speed and the flame combustion temperature were mainly investigated. Results show that the flame propagation speed and the flame combustion temperature increase continuously with the increase in preheating temperature, while it decreases continuously with the increase in CO<sub>2</sub> concentration. Additionally, the peak flame combustion temperature appears at an equivalent ratio of 1.1, while the peak flame propagation speed appears at an equivalent ratio of 2.0, which indicates that the flame combustion temperature is not the main factor affecting flame propagation speed for tail gas.

**Keywords:** tail gas, copper slag, CHEMKIN, flame propagation speed, flame combustion temperature

## 1 INTRODUCTION

Copper slag, a by-product of the copper smelting industry, contains a large amount of sensible heat and is composed of some metal oxides, which exhibit preferable catalytic performance in improving coal gasification (Du et al., 2010; Li et al., 2021). A heat recovery system from the heat of copper slag, which generates reducing tail gas via the endothermic reactions of gasification, was presented. The specific technology is shown in **Figure 1**. Copper slag itself is composed of a variety of metal oxides, so it has a certain catalytic reforming ability for low-molecular alkanes (He et al., 2002; Wang and Yu, 2018; Wang et al., 2021), and as a result, the composition of the tail gas changes greatly compared with the gas before reduction. The gas is mainly composed of H<sub>2</sub>, CO, CO<sub>2</sub>, and C<sub>n</sub>H<sub>m</sub>, which can be used as the reducing gas of copper slag for recycling valuable metals. While this reduction tail gas has a certain calorific value, it can also be used as fuel for gas turbines (Yalikunjiang-Tursun and Yue, 2019). Therefore, tail gas has received a particular interest in recent years.

Laminar flame propagation velocity is a basic parameter of fuel combustion, which can comprehensively reflect the exothermic capacity, diffusivity, and chemical reaction rate of the combustible gas mixture system (Hu and Yu, 2018; Shang et al., 2019). Laminar flame propagation velocity is not only an important basis for studying complex combustion but also an important parameter for verifying the chemical dynamic mechanism and guiding burner safety design (Dong et al., 2002; Bao and Liu, 2011). Flame temperature is also a basic parameter

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### Edited by:

Chungen Yin,  
Aalborg University, Denmark

### Reviewed by:

Yiheng Tong,  
China Space Foundation, China

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### Specialty section:

This article was submitted to  
Advanced Clean Fuel Technologies,  
a section of the journal  
Frontiers in Energy Research

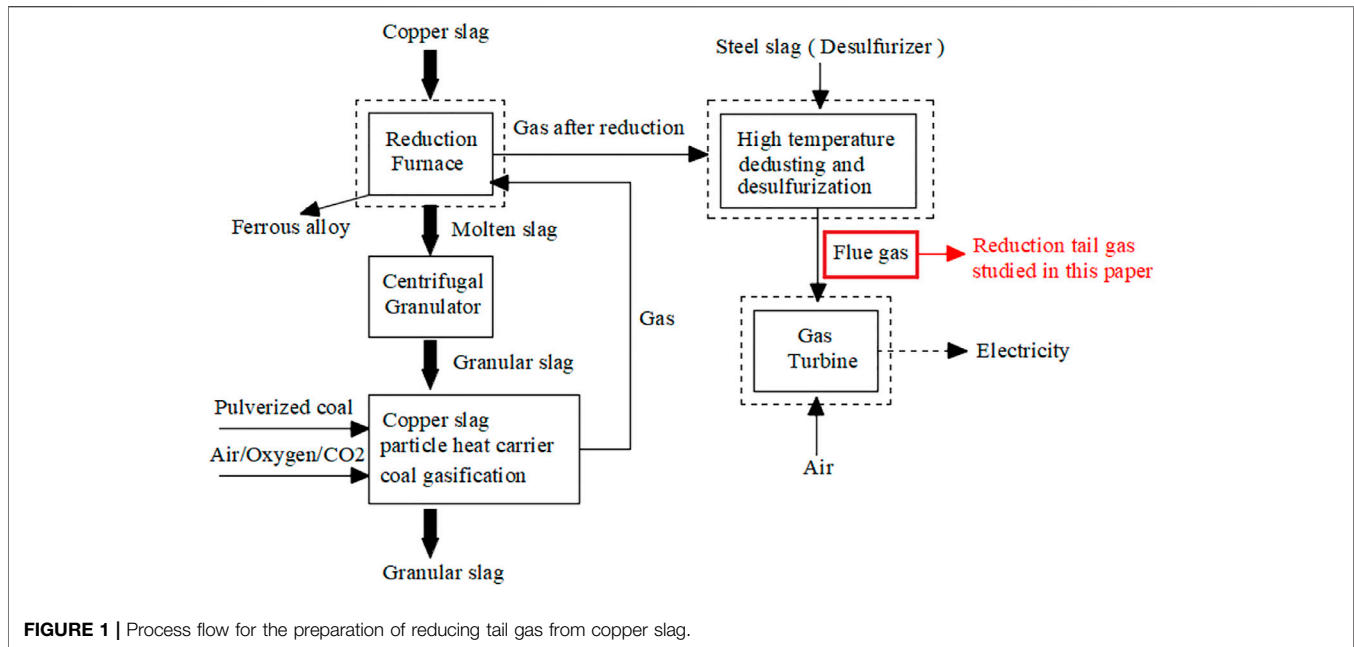
**Received:** 05 January 2022

**Accepted:** 08 March 2022

**Published:** 06 April 2022

### Citation:

Sun H, Xie H, Luo S, Zhang W, Zuo Z  
and Xianjun G (2022) Numerical  
Simulation of Flame Combustion  
Characteristics of Tail Gas From the  
Copper Slag Depleted Reduction.  
*Front. Energy Res.* 10:840383.  
doi: 10.3389/fenrg.2022.840383



of combustion and a key factor of overall efficiency of the combustion system, which affects all aspects of the material strength, structure, and design of the burner. Studying the combustion characteristics of laminar flame can help us better understand and interpret the actual combustion phenomenon together with the law of combustion reaction. Furthermore, understanding the laminar flame propagation mechanism is also the basis for the study of turbulent flame propagation mechanism (Kishore et al., 2008; Burluka et al., 2010; Zhang et al., 2010).

In recent years, domestic and foreign scholars have also done many experiments and model developments for laminar flame combustion characteristics. Among the study of the basic flame properties required for gas combustion, basic flame speed is very important because it provides detailed information on the thermochemical processes controlling fuel combustion (Bouvet et al., 2011; Zeng et al., 2019). A great deal of previous research into the laminar combustion velocity has focused on the Bunsen burner method (Lee et al., 2002; Pareja et al., 2009; Bouvet et al., 2011; Zhang et al., 2016), whose base-wood principle is to determine the laminar combustion velocity using a premixed flame cone. Research over the last decade indicates that laminar flame propagation velocity of gas can be affected by hydrogen volume fraction, CO<sub>2</sub> concentration, preheating temperature, and other factors (Han et al., 2016; Dai et al., 2020). In addition, some domestic and foreign scholars have used CHEMKIN software to simulate the flame combustion characteristics, and the simulation results are consistent with the results obtained by the basic experiments (Mueller et al., 2000; Natarajan et al., 2007, 2008; Ouimette and Seers, 2008; Krejci et al., 2013). At present, some studies have been carried out to simulate the flame propagation velocity of natural gas by using the raw lamp experiment method and CHEMKIN simulation software combined with GRI-3.0 reaction mechanism (Li et al., 2004). The results show that the

GRI-3.0 reaction mechanism is more suitable for the simulation of methane combustion characteristics.

In general, there have been plenty of studies on the combustion of single gas. The laws of laminar flame combustion characteristics of natural gas have been proved previously; however, the results cannot be used directly on the analysis of the tail gas, and the studies on the laminar flame combustion characteristics of the tail gas are seldom referred to.

This study aims to use the flame speed calculation module in PREMIX of CHEMKIN to simulate the combustion characteristics of the tail gas after sulfide removal, to mainly discuss the influence law of the equivalence ratio, gas preheating temperature, and the CO<sub>2</sub> concentration on the flame propagation speed and the flame combustion temperature. In this simulation, the reduced tail gas is generated from the endothermic gasification reacting in a heat recovery system. Therefore, this work will lay the foundation and provide basic data for the promotion and application of reducing gas combustion technology.

## 2 CHEMKIN SIMULATION METHOD

### 2.1 Reaction Model

CHEMKIN is a chemical reaction kinetics package, which simulates gas-phase reactions and surface reactions by using detailed primitive reactions. The PREMIX model can be divided into two modules: burner stabilized flame module (BSF) and flame speed calculation module (FSC). The BSF module is suitable for solving the flame temperature distribution curve and the material concentration curve when the mass flow of premixed gas is known. The FSC module is suitable for pressure constant, premixed gas

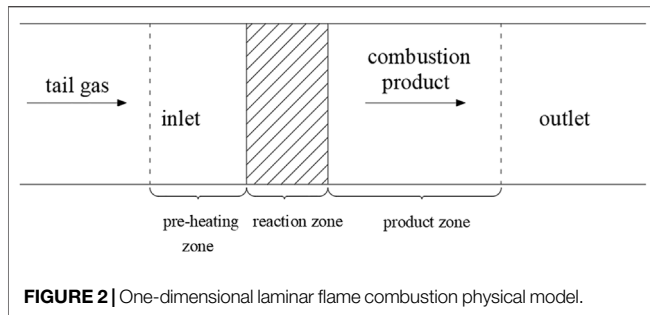


FIGURE 2 | One-dimensional laminar flame combustion physical model.

temperature, and known proportion. The flame temperature distribution and flame propagation velocity curve distribution can be conveniently obtained. In this study, the FSC module is used to simulate the combustion characteristics of the tail gas because the atmospheric pressure is constant and the gas temperature and ratio are known.

The one-dimensional laminar flame combustion physical model is shown in **Figure 2**. The tail gas from the copper slag depleted reduction enters the preheating zone from the inlet, and the gas temperature rises continuously, while the chemical reaction proceeds slowly. After the gas reaches the combustion temperature, it burns vigorously in the reaction zone, the chemical reaction proceeds violently, and the reaction reaches a dynamic equilibrium. The rise of gas temperature gradually slows down, and the final temperature tends to be certain in the product zone. Then, the chemical reaction rate drops rapidly, and the combustion products are finally discharged from the outlet.

## 2.2 Reaction Mechanism

All calculations in CHEMKIN software are based on the mechanism file. This is because the reaction of each primitive and its related parameters, the data of thermodynamic properties of each component, and the data of transport properties are included in the reaction mechanism file. The CO/H<sub>2</sub>/NO<sub>x</sub> mechanism file proposed by the National University of Ireland in collaboration with Xi'an Jiaotong University was used for this simulation, which is capable of simulating various reaction systems including H<sub>2</sub>/O<sub>2</sub>/N<sub>2</sub>, N<sub>2</sub>O pyrolysis, H<sub>2</sub>/N<sub>2</sub>O, H<sub>2</sub>/NO<sub>x</sub>/O<sub>2</sub>, CO/H<sub>2</sub>O/NO<sub>x</sub>/O<sub>2</sub>, and H<sub>2</sub>/CO/H<sub>2</sub>O/NO<sub>x</sub>/O<sub>2</sub> (Frassoldati et al., 2007; Zhang et al., 2017). Meanwhile, the mechanism contains a large amount of experimental data, including 104 excitation tube datasets, 87 JSR datasets, 87 FR datasets, and six laminar flame velocity datasets with more than 3,000 experimental points. In general, the mechanism file can simulate the combustion process of the tail gas well and has good simulation effect, so this mechanism file is chosen to simulate the combustion characteristics.

Assuming that the gas inlet is a one-dimensional steady flow, the governing equation can be simplified as follows:

Continuity:

$$\dot{M} = \rho uA \quad (1)$$

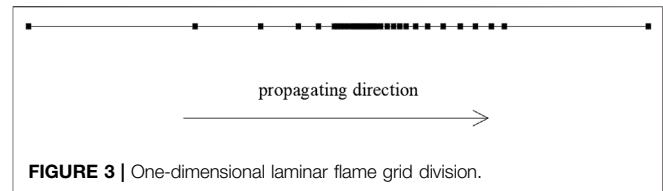


FIGURE 3 | One-dimensional laminar flame grid division.

Energy:

$$\dot{M} \frac{dT}{dx} - \frac{1}{C_p} \frac{d}{dx} \left( \lambda A \frac{dT}{dx} \right) + \frac{A}{C_p} \sum_{k=1}^k \rho Y_k V_k C_{pk} \frac{dT}{dx} + \frac{A}{C_p} \sum_{k=1}^k \dot{\omega}_k h_k W_k + \frac{A}{C_p} Q_{rad} = 0 \quad (2)$$

Species:

$$\dot{M} \frac{dY_k}{dx} + \frac{d}{dx} (\rho A Y_k V_k) - A \dot{\omega}_k W_k = 0, (k = 1, \dots, K_g) \quad (3)$$

Equation of state:

$$\rho = \frac{P\bar{W}}{RT}, \quad (4)$$

where  $x$  represents the one-dimensional coordinates,  $\dot{M}$  represents the total mass flow rate of gas components,  $Y_k$  denotes the  $k$ th species mass fraction,  $T$  represents the temperature in the  $x$  direction,  $P$  represents the pressure in the  $x$  direction,  $\rho$  means the density of the mixture,  $u$  represents the velocity of the mixed fluid in the  $x$  direction,  $W_k$  represents molecular weight of the  $k$ th species,  $R$  is the general constant of gas,  $\bar{W}$  represents the average molecular weight of the mixture,  $\lambda$  is the heat conductivity of the gas,  $C_p$  represents the heat capacity of the mixed gas at constant pressure,  $C_{pk}$  expresses the  $k$ th species' heat capacity at constant pressure,  $\dot{\omega}_k$  is the production molar rate through the chemical reaction of the  $k$ th species in every unit volume,  $V_k$  indicates the diffusion rate of the  $k$ th gas,  $h_k$  indicates the total enthalpy of the  $k$ th gas,  $Q_{rad}$  expresses the radiation heat loss of mixture, and  $A$  indicates the cross-sectional area of the flow tube.

## 2.3 Grid Model Establishment and Independence Test

According to the one-dimensional laminar flame physical model, the exhaust gas passes through the preheating zone, the reaction zone, and the product zone in turn. The path of the gas that passes through is established as a one-dimensional model, and the number of grids is divided. The number of grids is an important basis to determine the calculation accuracy and simulation time. If the number of grids is too small, the simulation accuracy is too poor, while if the number of grids is too large, the simulation calculation time is too long. Meanwhile, the temperature gradient and concentration gradient are smaller in the preheating zone and product zone but larger in the reaction zone. Therefore, the grid cannot be divided equally. According to the gradient, the number of grids in

**TABLE 1** | Test of optimal number of grids.

Grid quantity	Flame propagation speed (cm/s)	Changing range	Computing time (s)
100	Incalculability	-	-
150	343.228	0.105	33
200	343.123	0.111	37
250	343.012	0.215	40
300	342.797	0.074	45
500	342.723	-	53

**TABLE 2** | Main gas components of tail gas (volume fraction).

CO	H <sub>2</sub>	CO <sub>2</sub>	H <sub>2</sub> O
%	%	%	%
35.95	39.90	4.62	19.53

**TABLE 3** | Simulation calculation condition setting.

	Equivalence ratio	Preheating temperature (K)	CO <sub>2</sub> concentration (%)
1	1.0	598	4.62
2	0.6–3.0	598	4.62
3	1.0	298–798	4.62
4	1.0	598	4.62, 20, 30

the reaction zone is the densest, followed by the product zone, and the number of grids in the preheating zone is the least. The grid division of the laminar flame is shown in **Figure 3**.

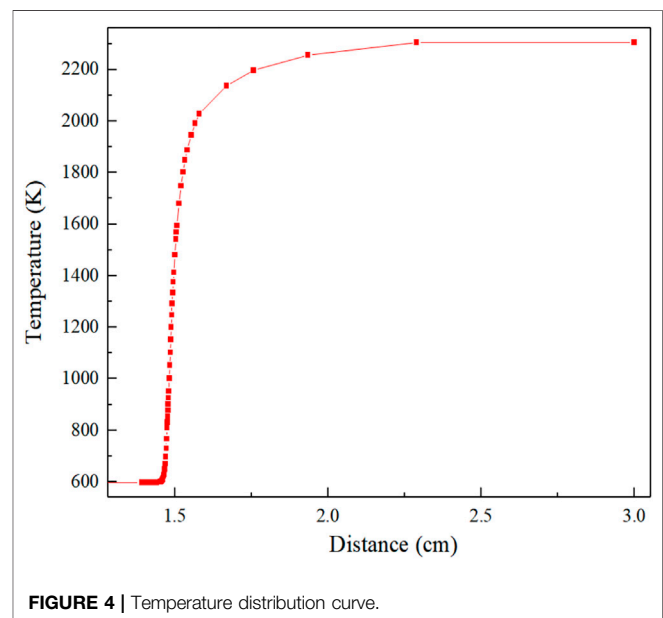
**Table 1** shows the results of the grid independence test of the tail gas under simulated conditions of pressure 1 atm, preheating temperature 598 K, and equivalence ratio 1.0 by using TWOPNT function self-meshing principle.

It can be concluded from the aforementioned table that when the number of grids changes from 100 to 300; the changing ranges of flame propagation speed are 0.105, 0.111, and 0.215, respectively; and the error is large. When the number of grids increases from 300 to 500, the changing range of flame propagation speed is 0.074 and the error is small. Therefore, when the number of grids is 300, the truncation error is small and the amount of computation is reduced, and the simulation accuracy and simulation time reach the best balance.

## 2.4 Materials and Calculation Conditions

The tail gas used in this simulation is generated by the endothermic reaction of copper slag gasification. The specific process technology is shown in **Figure 1**. **Table 2** shows the main gas composition of the tail gas used in this simulation.

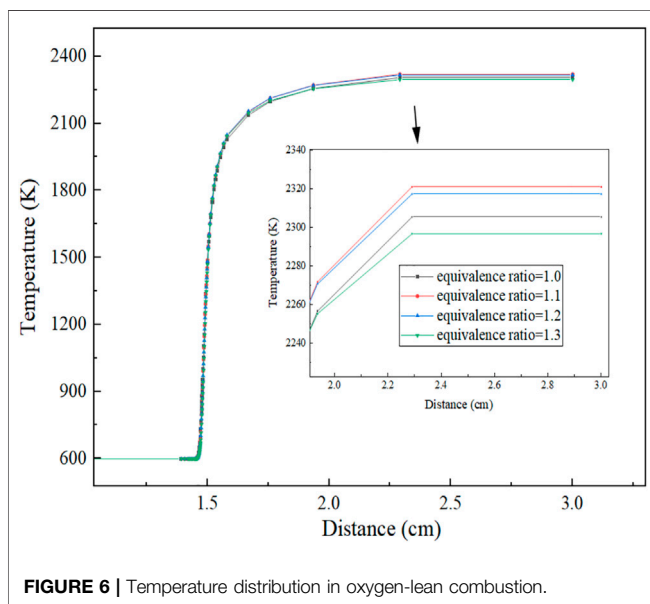
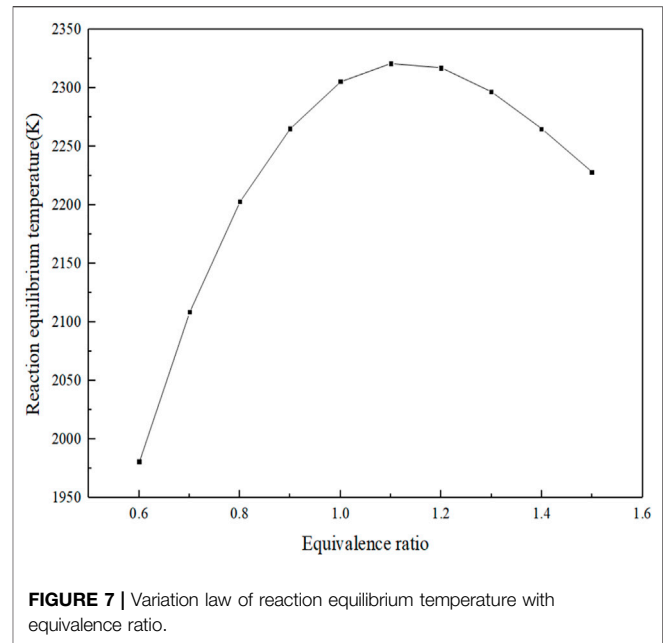
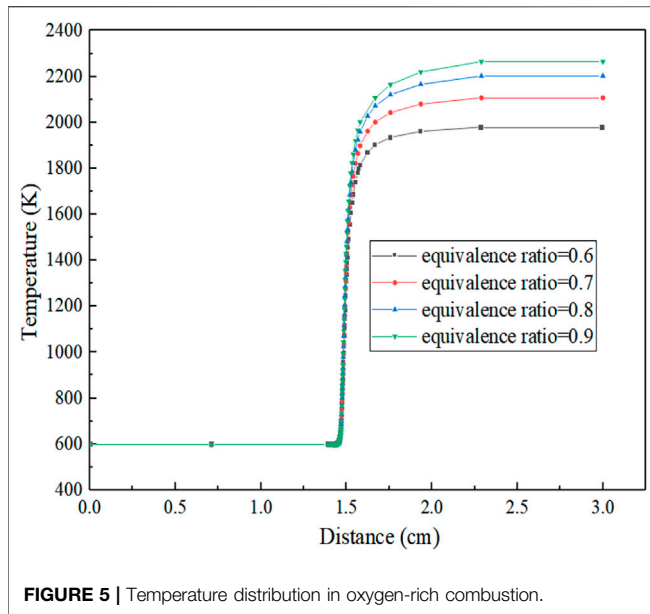
In this article, the effects of the equivalence ratio, preheating temperature, and the CO<sub>2</sub> concentration on the combustion characteristics of a one-dimensional laminar premixed flame were investigated. During the calculation, the pressure was kept at 1 atm, the premixed gas inlet velocity was set at 40 cm/s, and the ordinary air at 298 K, and 1 atm was used for calculation air. Other conditions for the simulation are shown in **Table 3**.



## 3 RESULTS AND DISCUSSION

### 3.1 Flame Temperature Distribution

**Figure 4** shows the flame temperature distribution curve. Based on the grid self-division function, the analysis is carried out with the curve of equivalent ratio = 1.0. The



preheating zone is 0–1.42 cm, and the flame temperature gradient is close to 0. The reaction zone is 1.42–1.66 cm. In this zone, intense heat and mass exchange occur, and when  $X = 1.66$  cm, the temperature reaches 88% of the complete reaction temperature (2305 K). After  $X = 1.66$  cm is the product region, in which the flame combustion temperature rises slowly, the heat–mass exchange gradually reaches equilibrium, and the chemical reaction through a longer product zone to reach equilibrium. The calculation results show that when  $X = 3.00$  cm, the temperature reaches stability and the maximum reaction temperature is 2169 K.

## 3.2 Simulation of Flame Combustion Characteristics at Different Equivalence Ratios

### 3.2.1 Effect of the Equivalence Ratio on the Flame Combustion Temperature

**Figure 5** shows the temperature distribution under oxygen-enriched combustion. It can be observed that the ranges of preheating zone, reaction zone, and exothermic zone are basically similar for different equivalence ratios, and there is little change in the distance. The equivalence ratio mainly affects the flame combustion temperature and flame temperature gradient. When the equivalence ratio is less than 1.0, the temperature gradient decreases with the decrease in the equivalence ratio, the flame temperature slowly increases with the increase of distance, and finally, the reaction equilibrium temperature gets lower and lower. This is because with the decrease in equivalence ratio, the amount of substances in the air is more than that of fuel, and finally, the air is in a surplus state. However, the lower the concentration of fuel, the slower the reaction speed and the slower the heat release. Therefore, in an adiabatic condition, the accumulated heat is lower, which leads to the lower and lower equilibrium reaction temperature at the macro-level.

As can be clearly seen from **Figure 6**, when the equivalent ratio is 1.1, the equilibrium reaction temperature is the highest, about 2321 K; when the equivalent ratio is less than 1.1, the reaction equilibrium temperature gradually decreases and the flame temperature rises slowly. The lower the air concentration and the lower the accumulated heat under adiabatic conditions, the lower the equilibrium reaction temperature at the macro-level. Calculations show that the temperature gradient is 0 when  $X = 3.0$  cm and the reaction temperature reaches equilibrium. The relationship between the specific reaction equilibrium

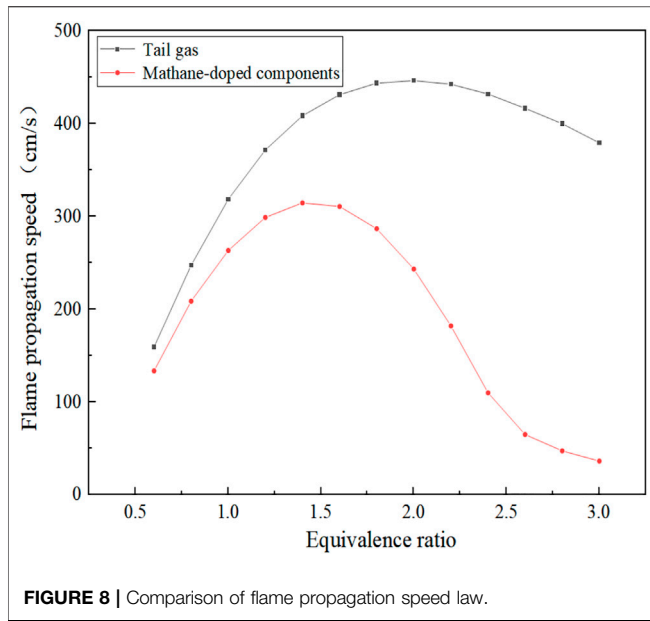


FIGURE 8 | Comparison of flame propagation speed law.

temperature and the equivalence ratio is shown in **Figure 7**, from which it can be clearly seen that when the equivalence ratio is 1.1, the equilibrium reaction temperature reaches the maximum.

### 3.2.2 Effect of the Equivalence Ratio on the Flame Propagation Speed

**Figure 8** shows the comparison of the tail gas with a typical methane flame propagation speed law. It can be seen that the flame propagation speed increases at first with the increase in the equivalence ratio and then decreases. When the equivalence ratio changes from 0.6 to 2.0, the flame propagation speed rises rapidly; while in the range of 2.0–3.0, the flame propagation speed decreases slowly. As the equivalence ratio changed to 2.1, the flame propagation speed achieved a maximum speed of 446.58 cm/s. According to the variation trend of flame propagation velocity of typical methane component, the maximum flame propagation velocity is achieved at the equivalence ratio of 1.4 after the methane component is mixed in the fuel component. Compared with the fuel tail gas, the flame propagation velocity rises at the equivalence ratio of 0.6–1.4 while it falls at the equivalence ratio of 1.4–3.0. In addition, the gradient of the falling phase is significantly larger than the gradient of the rising phase, that is, the flame propagation velocity decreases extremely fast, and the maximum value of flame propagation velocity is smaller than that of the fuel components in this study.

The reason for the aforementioned differences mainly lies in the composition of the fuel components, when the fuel contains a certain amount of  $\text{CH}_4$ , and the equivalence ratio corresponding to the maximum flame propagation velocity moves from the fuel-rich region to 1.0. From the point of view of chemical kinetics, the increase in laminar flame propagation speed has a strong correlation with the maximum value of free radicals such as H, O, and OH. When  $\text{CH}_4$  is added, the reaction rate of the chain branching reaction ( $\text{H} + \text{O}_2 = \text{O} + \text{OH}$ ) with the largest generation rate of OH radicals decreases, while the reaction

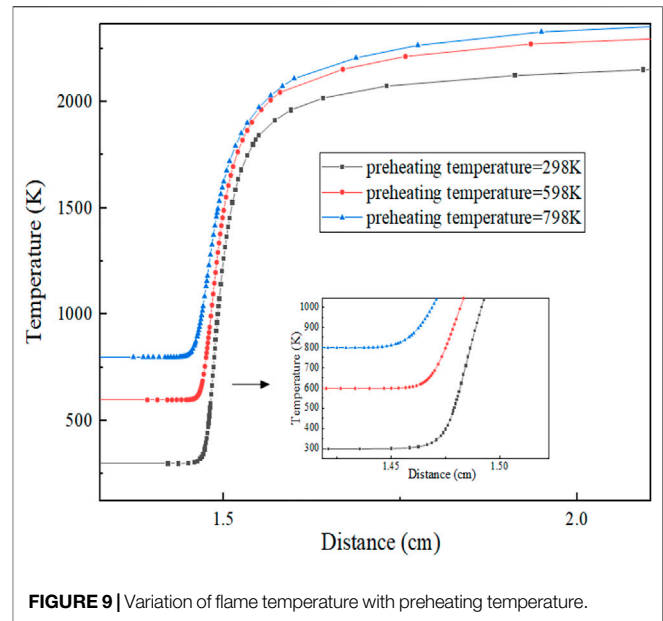


FIGURE 9 | Variation of flame temperature with preheating temperature.

rate of the main oxidation reaction ( $\text{CO} + \text{OH} = \text{CO}_2 + \text{H}$ ) decreases, and the sensitivity coefficient also decreases. At the same time,  $\text{CH}_4$  and  $\text{CH}_3$  consume highly active H and OH, resulting in the decrease in laminar flame propagation speed. As the experimental results obtained by Liu Jiaqi et al. using the Bunsen burner method, the equivalence ratios of the maximum flame propagation velocities were 1.3, 1.2, and 1.1 for 2, 5, and 8% of the methane blending mass, respectively. The gas is affected by the thermodynamic dilution effect of  $\text{H}_2\text{O}$ , resulting in the aforementioned phenomenon (Liu, 2018). After the addition of  $\text{CH}_4$ , the molar share of  $\text{H}_2$  decreases, which makes the decrease in flame propagation speed accordingly (Xiao et al., 2020). As can be seen from **Figure 7**, the peak adiabatic combustion temperature occurs at an equivalence ratio of 1.1, while the peak flame propagation velocity occurs at an equivalence ratio of 2.0. Therefore, for the fuel components in this article, the adiabatic combustion temperature is not the main factor affecting the flame propagation velocity.

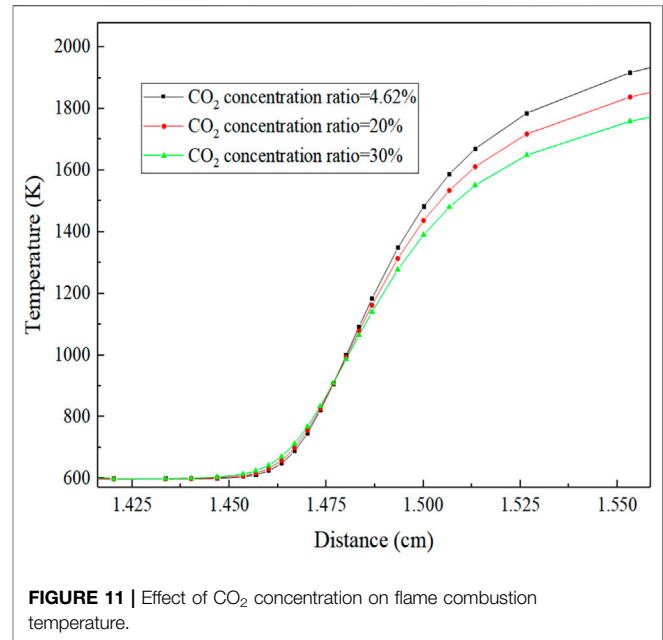
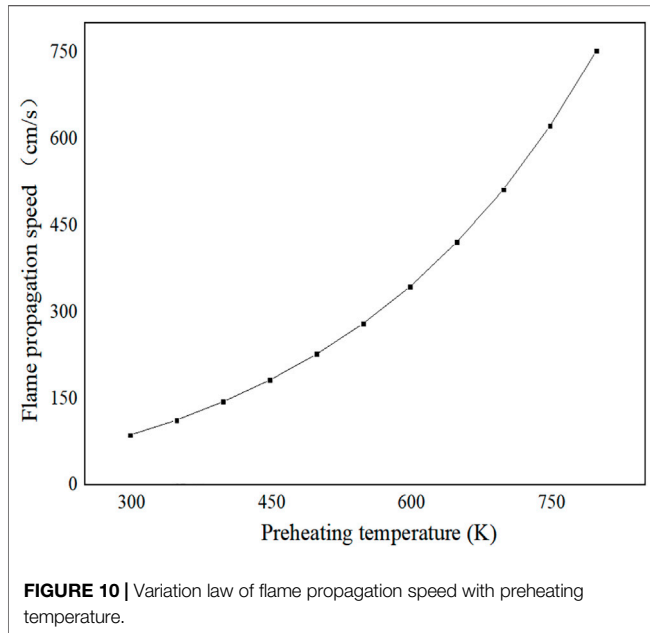
## 3.3 Simulation of Flame Combustion Characteristics at Different Preheating Temperatures

### 3.3.1 Effect of the Preheating Temperature on the Flame Combustion Temperature

**Figure 9** shows the variation of flame temperature with preheating temperature. It can be seen from the figure that the reaction zone decreases with the increase in preheating temperature, while the reaction temperature rises with the increase in preheating temperature, and the specific reaction zones are shown in **Table 4**. The reason for the aforementioned phenomenon is that with the increase in preheating temperature, the number of activating molecules increases, the activation energy required for the reaction decreases, and the basic reaction becomes more intense.

**TABLE 4** | Relationship between preheating temperature and reaction area size.

Preheating temperature (K)	Reaction region (cm)	Reaction equilibrium temperature(K)
298	1.4668–1.7310	2,169
598	1.4601–1.6690	2,321
798	1.4480–1.5998	2,383



Finally, at the macro-level, the higher the equilibrium temperature of the reaction, the larger the temperature gradient and the smaller the reaction area.

It can be seen from **Figure 9** that with the increase in preheating temperature, the starting position of the reaction zone is approaching coordinate zero. This is because the increase in temperature increases the number of activated molecules in the preheating zone and the probability and intensity of collisions between molecules, so it is easier to start the reaction. And the closer the starting position of the reaction zone is to the coordinate zero. In other words, the preheating zone decreases, while the product zone increases.

### 3.3.2 Effect of the Preheating Temperature on the Flame Propagation Speed

The influence of preheating temperature on laminar flame velocity can be described by the following formula:

$$S_u \propto \sqrt{\frac{\bar{\alpha}}{\rho}} \exp\left(-\frac{E\alpha}{RT_{rz}}\right), \quad (5)$$

where  $\alpha$  is the diffusion coefficient of the mixture,  $\rho$  means the density of the mixture, and  $T_{rz}$  is the temperature of the chemical reaction zone. The increase in preheating temperature will increase the temperature of the chemical reaction zone and

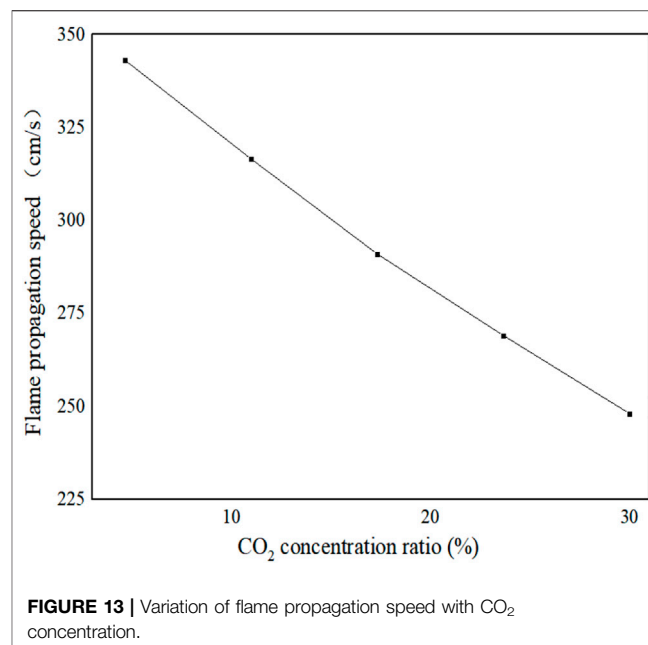
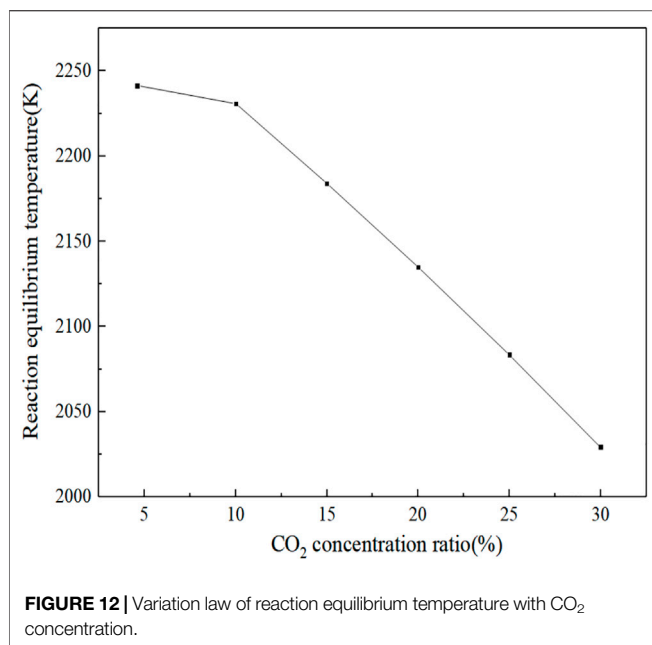
reduce the density of the gas, but the increase in density will be small. The increase in preheating temperature mainly affects the laminar flame velocity by affecting the temperature of the chemical reaction zone in the exponential term. Therefore, it is theoretically inferred that the laminar flame velocity increases exponentially with the increase in preheating temperature.

**Figure 10** shows the variation law of flame propagation speed with the preheating temperature. From **Figure 10**, it can be concluded that the flame propagation element is directly proportional to the preheating temperature. The reason is that the increase in preheating temperature increases the number of active molecules and reduces the activation energy of the reaction. Therefore, the reaction can proceed quickly and the heat as well as the mass exchange is violent, resulting in a faster flame propagation.

## 3.4 Simulation of Flame Combustion Characteristics at Different CO<sub>2</sub> Concentrations

### 3.4.1 Effect of CO<sub>2</sub> Concentration on the Flame Combustion Temperature

**Figure 11** shows the law of the influence of CO<sub>2</sub> concentration on the flame burning rate. As the gas enters the reaction zone, the flame burning temperature starts to change. In the range of



1.4468–1.4767 cm, the flame burning rate increases with the increase in CO<sub>2</sub> concentration, and when  $X > 1.4767$  cm, the flame burning temperature decreases with the increase in CO<sub>2</sub> concentration. The final equilibrium temperature of the reaction decreases with the increase in carbon dioxide concentration, as shown in **Figure 12**. The reaction equilibrium temperature decreases linearly with CO<sub>2</sub>. The reason is that CO<sub>2</sub> is a non-combustible component. When combustible components burn and release heat, CO<sub>2</sub> will absorb the heat released in the reaction process but will not spontaneously burn and release heat. Therefore, the higher the concentration of CO<sub>2</sub>, the lower the flame combustion temperature and even the non-combustible situation will occur.

### 3.4.2 Effect of the CO<sub>2</sub> Concentration on the Flame Propagation Speed

The thermodynamic effect of CO<sub>2</sub> on the flame velocity is mainly to reduce the total amount of reactants and the overall heat release, increase the specific heat capacity of the mixture, reduce the flame temperature, and change the transport properties of the mixture. But CO<sub>2</sub> is not just an inert diluent; on the contrary, it can participate in chemical reactions.

**Figure 13** shows the variation of flame propagation velocity with CO<sub>2</sub> concentration. It can be seen from **Figure 13** that the flame propagation velocity decreases with the increase in CO<sub>2</sub> concentration. This is because when the CO<sub>2</sub> component is involved in the chemical reaction, the suppression of flame propagation speed by CO<sub>2</sub> component mainly comes from two aspects: First, the thermodynamic properties of CO<sub>2</sub>; when the CO<sub>2</sub> is used as a fuel component, it is non-combustible, thus causing an overall reduction in fuel heat generation, which increases the specific heat capacity of the resulting mixture as well as lowers the adiabatic temperature, to suppress the flame propagation rate. Second, the chemical effect of CO<sub>2</sub>, unlike the

typical inert component N<sub>2</sub>, is directly involved in the process of radical reaction, for instance,  $\text{CO} + \text{OH} = \text{CO}_2 + \text{H}$ , which inhibits the oxidative combustion reaction of CO but also competes with the branching radical reaction  $\text{H} + \text{O}_2 = \text{O} + \text{OH}$  for H. In this way, active free radicals such as O, OH, and H are reduced, thereby reducing the flame propagation speed, which is consistent with the results of Han and Zhang (Zhang et al., 2011; Han et al., 2016).

## 4 CONCLUSION

In this study, the thermal properties of combustion of the reduced tail gas were simulated by CHEMKIN, and the basic parameters of flame combustion were studied. The main findings have been summarized as follows:

- (1) The flame propagation speed and the flame combustion temperature increase continuously with the increase in preheating temperature, while they decrease continuously with the increase in CO<sub>2</sub> concentration.
- (2) Compared with the typical methane flame propagation velocity distribution, the results show that the gradient of the decreasing phase of flame propagation velocity of the tail gas is significantly lower than that of methane flame. However, the maximum value of flame propagation velocity of the tail gas is higher than that of a typical methane flame, indicating that the flame propagation velocity decreased due to the addition of CH<sub>4</sub> for the tail gas component.
- (3) The peak flame combustion temperature appears at an equivalent ratio of 1.1, while the peak flame propagation speed appears at an equivalent ratio of 2.0, which indicates that the flame combustion temperature is not the main factor affecting flame propagation speed for the tail gas.



## DATA AVAILABILITY STATEMENT

The original contributions presented in the study are included in the article/Supplementary Material, further inquiries can be directed to the corresponding author.

## AUTHOR CONTRIBUTIONS

HS performed the data analysis and wrote the manuscript. HX performed the simulations using software. SL provided ideas and directed the analysis of the data. WZ and ZZ provided

constructive comments on revisions to the manuscript. XG interpreted the results.

## FUNDING

This work supported by 2020 science and technology project of Qingdao West Coast New Area (Science and technology benefiting the people) (2020-99) and the National Key Research and Development of China (2020YFD1100302).

## REFERENCES

- Bao, X., and Liu, F. (2011). Experimental Measurement and Simulation Calculation of Laminar Combustion Velocity of Hydrogen/air Mixture. *Combustion Sci. Technology* 17 (05), 407–413. CNKI:SUN:RSKX.0.2011-05-004.
- Bouvet, N., Chauveau, C., Gökalp, I., and Halter, F. (2011). Experimental Studies of the Fundamental Flame Speeds of Syngas (H<sub>2</sub>/CO)/air Mixtures. *Proc. Combustion Inst.* 33 (1), 913–920. doi:10.1016/j.proci.2010.05.088
- Burluka, A. A., Harker, M., Osman, H., Sheppard, C. G. W., and Konnov, A. A. (2010). Laminar Burning Velocities of Three C<sub>3</sub>H<sub>6</sub>O Isomers at Atmospheric Pressure. *Fuel* 89 (10), 2864–2872. doi:10.1016/j.fuel.2010.02.004
- Dai, L., Wang, K., Xu, P., and Hu, X. (2020). Experimental Study of Laminar Flame Propagation Velocity of CH<sub>4</sub>/H<sub>2</sub> Mixture in O<sub>2</sub>/CO<sub>2</sub> Atmosphere. *Metallurgical Energy* 39 (03), 29–32+64. CNKI:SUN:YJLY.0.2020-03-007.
- Dong, G., Liu, H., and Chen, Y. (2002). Semi-detailed Chemical Kinetic Mechanism of Generalized Methane Laminar Flow Premixed Flame. *Combustion Sci. Technology* 8 (01), 44–48. doi:10.3321/j.issn:1006-8740.2002.01.011
- Du, W., Yu, Q., and Li, P. (2010). “Research Progress of Chemical Sensible Heat Recovery Technology of Blast Furnace Slag,” in *2010 National Academic Conference on Energy and Thermal Engineering*, 5.
- Frassoldati, A., Faravelli, T., and Ranzi, E. (2007). The Ignition, Combustion and Flame Structure of Carbon Monoxide/hydrogen Mixtures. Note 1: Detailed Kinetic Modeling of Syngas Combustion Also in Presence of Nitrogen Compounds. *Int. J. Hydrogen Energ.* 32 (15), 3471–3485. doi:10.1016/j.ijhydene.2007.01.011
- Han, M., Ai, Y., Chen, Z., Wang, Y., and Kong, W. (2016). Laminar Flame Propagation Characteristics of H<sub>2</sub>/CO/O<sub>2</sub>/CO<sub>2</sub> at Different Preheating Temperatures. *J. Eng. Thermophys.* 37 (01), 189–193. CNKI:SUN:GCRB.0.2016-01-041.
- He, P., Yu, Y., Guo, S., Zhang, X., Jia, J., and Yin, C. (2002). Combustion Test in the Lifting Section of Solid Heat Carrier Coal Gasification Pilot Plant. *Metallurgical Energy* 21 (03), 52–55. doi:10.3969/j.issn.1001-1617.2002.03.015
- Hu, X., and Yu, Q. (2018). Effect of the Elevated Initial Temperature on the Laminar Flame Speeds of Oxy-Methane Mixtures. *Energy* 147, 876–883. doi:10.1016/j.energy.2018.01.088
- Krejci, M. C., Mathieu, O., Vissotski, A. J., Ravi, S., Sikes, T. G., Petersen, E. L., et al. (2013). Laminar Flame Speed and Ignition Delay Time Data for the Kinetic Modeling of Hydrogen and Syngas Fuel Blends. *J. Eng. Gas Turbines Power* 135 (2). doi:10.1115/1.4007737
- Lee, C.-E., Oh, C.-B., Jung, I.-S., and Park, J. (2002). A Study on the Determination of Burning Velocities of LFG and LFG-Mixed Fuels. *Fuel* 81 (13). doi:10.1016/S0016-2361(02)00049-2
- Li, J., Zhao, Z., Kazakov, A., and Dryer, F. L. (2004). An Updated Comprehensive Kinetic Model of Hydrogen Combustion. *Int. J. Chem. Kinet.* 36 (10), 566–575. doi:10.1002/kin.20026
- Li, X., Dou, Z., Zhang, T., and Liu, Y. (2021). Progress of Comprehensive Utilization of Copper Smelting Slag. *Nonferrous Met. (Smelting Part)* (04), 108–118. doi:10.3969/j.issn.1007-7545.2021.04.018
- Liu, J. (2018). *Study on CO/H<sub>2</sub>/CH<sub>4</sub> Laminar Flame Propagation Velocity under O<sub>2</sub>/H<sub>2</sub>O Condition*. Harbin Municipality: Harbin Institute of Technology. master's degree.
- Mueller, M. A., Gatto, J. L., Yetter, R. A., and Dryer, F. L. (2000). Hydrogen/nitrogen Dioxide Kinetics: Derived Rate Data for the Reaction H<sub>2</sub> + NO<sub>2</sub> = HONO + H at 833 K. *Combustion and Flame* 120 (4). doi:10.1016/S0010-2180(99)00118-2
- Natarajan, J., Lieuwen, T., and Seitzman, J. (2008). Laminar Flame Speeds and Strain Sensitivities of Mixtures of H<sub>2</sub>/O<sub>2</sub>/N<sub>2</sub> at Elevated Preheat Temperatures. *J. Eng. Gas Turbines Power* 130 (6). doi:10.1115/1.2943180
- Natarajan, J., Lieuwen, T., and Seitzman, J. (2007). Laminar Flame Speeds of H<sub>2</sub>/CO Mixtures: Effect of CO<sub>2</sub> Dilution, Preheat Temperature, and Pressure. *Combustion and Flame* 151 (1), 104–119. doi:10.1016/j.combustflame.2007.05.003
- Ouimette, P., and Seers, P. (2009). Numerical Comparison of Premixed Laminar Flame Velocity of Methane and wood Syngas. *Fuel* 88 (3), 528–533. doi:10.1016/j.fuel.2008.10.008
- Pareja, J., Burbano, H. J., and Ogami, Y. (2010). Measurements of the Laminar Burning Velocity of Hydrogen-Air Premixed Flames. *Int. J. Hydrogen Energ.* 35 (4), 1812–1818. doi:10.1016/j.ijhydene.2009.12.031
- Ratna Kishore, V., Duhan, N., Ravi, M. R., and Ray, A. (2008). Measurement of Adiabatic Burning Velocity in Natural Gas-like Mixtures. *Exp. Therm. Fluid Sci.* 33 (1), 10–16. doi:10.1016/j.expthermflusci.2008.06.001
- Shang, R., Yang, Y., Gao, J., and Li, G. (2019). Experimental Study of Flame Propagation Velocity of Hydrogen-Doped Natural Gas Laminar Flow. *Chin. J. Saf. Sci.* 29 (11), 103–108. doi:10.16265/j.cnki.issn1003-3033.2019.11.017
- Wang, H., Fan, F., Li, P., and Pan, S. (2021). Analysis of Modern Coal Gasification Technology Progress and Industrial Status. *Coal Chem. Industry* 49 (04), 52–56. doi:10.19889/j.cnki.10059598.2021.04.012
- Wang, H., and Yu, Q. (2018). Experimental Study of Self-Heating Hot Carrier Coal Gasification. *Metallurgical Energy* 37 (06), 19–24. doi:10.3969/j.issn.1001-1617.2018.06.005
- Xiao, P., Lee, C.-f., Wu, H., and Liu, F. (2020). Effects of Hydrogen Addition on the Laminar Methanol-Air Flame under Different Initial Temperatures. *Renew. Energ.* 154, 209–222. doi:10.1016/j.renene.2020.03.037
- Yalikunjiang-Tursun, A.-A., and Yue, P. (2019). Characterization of Coal Gasification with Pyrolysis-Reforming-Combustion Decoupling. *J. Chem. Eng.* 70 (08), 3040–3049.
- Zeng, W., Liu, J., Ma, H., Liu, A., Liu, K., Liang, S., et al. (2019). Effect of Nitrogen Dilution Gas on the Combustion Characteristics of Natural Gas. *Therm. Sci. Technology* 18 (03), 219–227. doi:10.13738/j.issn.1671-8097.018098
- Zhang, H., Bai, X., Jeong, D., Cho, G., Choi, S., and Lee, J. (2010). Fuel Combustion Test in Constant Volume Combustion Chamber with Built-in Adaptor. *Sci. China Technol. Sci.* 53 (4), 1000–1007. doi:10.1007/s11431-009-0407-7
- Zhang, W., Chen, Z., and Kong, W. (2012). Effects of Diluents on the Ignition of Premixed H<sub>2</sub>/air Mixtures. *Combustion and Flame* 159 (1), 151–160. doi:10.1016/j.combustflame.2011.05.017

- Zhang, Y., Lu, H., Wang, Q., Yu, S., and Gao, W. (2016). Simulation and Experimental Testing of Natural Gas normal Flame Propagation Velocity. *Gas Heat* 36 (05), 17–22+33. doi:10.13608/j.cnki.1000-4416.2016.05.005
- Zhang, Y., Mathieu, O., Petersen, E. L., Bourque, G., and Curran, H. J. (2017). Assessing the Predictions of a NO X Kinetic Mechanism on Recent Hydrogen and Syngas Experimental Data. *Combustion and Flame* 182, 122–141. doi:10.1016/j.combustflame.2017.03.019

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