



Impact Localization in Composites Using Time Reversal, Embedded PZT Transducers, and Topological Algorithms

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Coles A, de Castro BA, Andreades C, Baptista FG, Meo M and Ciampa F (2020) Impact Localization in Composites Using Time Reversal, Embedded PZT Transducers, and Topological Algorithms. Front. Built Environ. 6:27. doi: 10.3389/fbuil.2020.00027 Time reversal is a powerful imaging processing technique that focuses waves at their original source using a single receiver transducer when diffusive wave field conditions are met. This has been successfully proved on various engineering components and materials using elastic waves with surface bonded transducers. This paper investigates the performance of time reversal for the localization of impact sources on fiber reinforced plastic composite structures with embedded piezoelectric sensors. A topologic approach, here named as minimum average method, is proposed to enhance the accuracy of time reversal in retrieving the impact location. Experimental tests were carried out to validate the robustness and reliability of time reversal against traditional topological approaches by altering impulsive responses contained in the baseline signals. Impact localization results revealed that time reversal and the new topological approach provided high accuracy in identifying the impact location, particularly in the presence of double impacts and material damage, which were not accounted during the initial training process. Results indicate that time reversal with embedded transducers has potential to be effective in real operating conditions, where alterations of acoustic emission responses in the baseline signals are less predictable.

Keywords: time reversal signal processing, impact localization, embedded sensors, piezoelectric transducers, composite plates

INTRODUCTION

Fiber reinforced plastic (FRP) composite materials are becoming increasingly popular, particularly in the aerospace field, due to their higher mechanical benefits over metals such as low weight and high stiffness and strength (Chandrashekhar and Ganguli, 2016; Baccar and Söffker, 2017; Castro et al., 2019). However, composite components can often demonstrate fragility toward low-velocity impacts (Atobe et al., 2011; Marchi et al., 2011; Ciampa et al., 2016; Ali et al., 2017), which can considerably degrade the structural integrity and, if not detected, they can result in catastrophic failures. Acoustic emission (AE)-based structural health monitoring (SHM) systems have become an important tool to provide in-service detection and localization of low-velocity impacts, thus enhancing safety of composite components. Literature presents various AE localization methods developed over the years (Haider et al., 2018; Morse et al., 2018; Qiao et al., 2019; Qiu et al., 2019; Tian et al., 2019; Yin et al., 2019). However, only few AE source identification systems can be used in anisotropic inhomogeneous composite materials, especially if no prior knowledge of the propagating wave field and mechanical properties of the host component are provided (Tobias, 1976; Ciampa and Meo, 2010a; McLaskey, 2010), which is generally common in practical aerospace applications.

SHM algorithms for AE localization in composite materials typically rely on the time of arrival (TOA) estimation (Ciampa and Meo, 2010b; Ciampa et al., 2012; Kundu, 2012; Kundu et al., 2012; Emanuele et al., 2017; Seno et al., 2019). However, TOA-based techniques require a relatively large number of receiver transducers measuring the coherent part of the wave field (ballistic wave). Advanced signal processing methods such as peak detection (Castro et al., 2018; Das and Leung, 2018; Zhou et al., 2019), cross-correlation (Kim et al., 2015; Li et al., 2018), Hilbert (Chen et al., 2018; Kim and Yuan, 2018), and wavelet transformation (Bianchi et al., 2015; Zhu et al., 2017) have been employed to extract physical parameters from measured data in order detect AE waves. Nevertheless, the dispersive nature of guided waves, as well as boundary reflections and mode conversion in geometrically complex components can all alter the acquired signal resulting in incorrect TOA estimations (Ciampa and Meo, 2011). Ciampa and Meo (2011) recently developed a time reversal (TR) signal processing technique to compensate the dispersive behavior of guided waves and to enhance the localization (imaging) process in inhomogeneous FRP composites. TR method requires, in principle, only one receiver transducer and, unlike current TOAbased AE monitoring systems, it uses a diffusive wave field to achieve focusing of the source with high accuracy. TR process is typically divided into two steps. The first one involves training of the structure under investigation by creating a baseline grid in which each point pertains to a corresponding recorded impulse response (i.e., the experimental Green's function). The second step consists of correlating the impulsive transfer function associated to each baseline grid point against the inversion of the structural response of a new impact of unknown location. The predicted impact cell is identified as the cell with the greatest correlation. Within each cell, a more precise location can be achieved with topographic methods such as the center of gravity (CoG) technique.

TR has been tested on both isotropic plate-like and complex composite components with various experimental conditions (Ing and Quieffin, 2005; Chen and Yuan, 2010; Bas et al., 2011; Qiu et al., 2011; Chen et al., 2012a; Park et al., 2012; Remillieux et al., 2015). For example, Ciampa and Meo (2014) used TR on a composite tail rotor blade of a helicopter made of both glass and carbon fiber materials with varying thickness. Experimental results demonstrated that the calculated impact location could be achieved with a high level of accuracy at any point on the structure. The same authors (Ciampa et al., 2016) tested the effects of TR by varying the number of excitation points in the baseline grid, the number of piezoelectric sensors and by compensating temperature changes using a forward step signal stretch (FSSS) technique. Although only one sensor is required for TR, it was shown that increasing the number of sensors and taking an average from the contribution decreases the probability of localization error (Ciampa et al., 2016).

In all TR tests performed to date on composite components, receiver piezoelectric (PZT) transducers have all been surface mounted (e.g., Ing and Quieffin, 2005; Chen and Yuan, 2010; Bas et al., 2011). However, in real operating conditions, external exposure of the sensor can lead to failure due to environmental factors such as moisture and corrosion (Birman, 1996). As a result, it is often desirable to embed transducers, thus creating "smart" FRP composite structures. The novelty of this paper is to evaluate, for the first time, the performance of TR for impact localization on smart composite structures with embedded receiver PZT sensors. For the electrical insulation of PZTs into carbon/epoxy composites, a new manufacturing process developed by Andreades et al. (2018, 2019) was used. It consists of using thin E-glass fiber fabric patches as insulating interlayers for PZT sensors, which provide enhanced adhesion with the surrounding epoxy matrix against standard insulation techniques adopting Kapton and Teflon patches (Huang and Nemat-Nasser, 2007; Clausen et al., 2013; Andreades et al., 2018). Another innovative aspect of this paper is to further enhance the reliability and accuracy of the TR imaging process by proposing a new topological algorithm for impact localization, here named as minimum average (MA), and comparing its performance against the CoG technique. Various case studies were analyzed to simulate real operational conditions, including changes of the training (baseline) signals and double impacts, for which the alteration to AE responses are less predictable. The outline of this research work is as follows: in section Time Reversal Theory, the time reversal impact localization algorithm is introduced. The new topological MA approach and the traditional CoG method are described in section Topological Approaches for Impact Localization. Section Experimental Set-Up shows the set-up used to perform experimental tests, whilst in section Impact Localisation Results are illustrated. Conclusions are reported in section Conclusion.

TIME REVERSAL THEORY

The TR method is conventionally split into two steps: a "forward" and a "backward" propagation step (Chen et al., 2012b; Ciampa and Meo, 2012, 2014; De Simone et al., 2018). It also used for guided waves applications (Wang et al., 2004; Huang et al., 2019) but, in this work, the time reversal process is applied as a step of a new passive impact localization technique. According to (De Simone et al., 2018), during the "forward" step the structure is divided into a grid containing M excitation points. Each point is, in turn, excited by an impulsive waveform such as that caused by a low velocity impact. N passive sensors record the local response for each of the M impacts, resulting in a data set of size $M \times N$. Using the hypothesis of free unbounded space, assuming the wave field $\Psi(\mathbf{r}, t)$ can be measured at any point

over the structure and all deformations are linearly elastic, the general solution of the elastodynamic wave equation can be simplified to

$$\Psi(\mathbf{r},t) = \iiint_{\Omega} \left[G(\mathbf{r},t_0;\mathbf{r}_0) \bigotimes_{t} e(\mathbf{r}_0,t) \right] d\Omega_0$$
(1)

where $\mathbf{r} = x_1\hat{i} + x_2\hat{j} + x_3\hat{k}$ in Lagrangian coordinates, t represents time, $G(\mathbf{r}, t_0; \mathbf{r}_0)$ is the Green space-time function, showing the wave field produced at \mathbf{r} at time t by an impulsive impact located at \mathbf{r}_0 excited at time t_0 . The symbol \bigotimes represents the convolution over time, Ω is the solid volume and $e(\mathbf{r}_0, t)$ is the excitation function indicating an impulsive force (Ciampa and Meo, 2014). Due to the impacts behaving as impulse responses, it is reasonable to treat the excitation function as a Dirac delta function. In this case the Green space-time function is called "impulse response" and is equal to the measured wave field (De Simone et al., 2018). The excitation points are used to convert the surface of the structure into a discrete domain with M points at various locations \mathbf{r}_m , meaning the spatial integral in Equation (1) becomes a sum of convoluted components,

$$\Psi(\mathbf{r},t) = \sum_{m=1}^{M} G(\mathbf{r},t_{0};\mathbf{r}_{0}) \underset{t}{\otimes} e(\mathbf{r}_{0},t).$$
(2)

The "backward" propagation step involves the correlation of the waveform emitted at an unknown location r_{m0} with the impulse responses of each excitation point. It can be shown (Ciampa and Meo, 2014; De Simone et al., 2018) that

$$R_{TR} = G(\mathbf{r}_m, t_0; \mathbf{r}) \bigotimes_t G(\mathbf{r}, -t; \mathbf{r}_{m0})$$

=
$$\int_0^t G(\mathbf{r}_m, t; \mathbf{r}) G(\mathbf{r}, t + \tau; \mathbf{r}_{m0}) d\tau \qquad (3)$$

is maximum when $\mathbf{r}_m = \mathbf{r}_{m0}$, where τ is the time lag and R_{TR} is the TR operator equal to the cross-correlation between the baseline impulse response and the impulse response of the new impact of unknown location. The Cauchy-Schwarz inequality (Aldaz et al., 2015) proves that

$$|R_{TR}|^{2} = \left| \int_{0}^{t} G(\mathbf{r}_{m}, t; \mathbf{r}) G(\mathbf{r}, t + \tau; \mathbf{r}_{m0}) d\tau \right|^{2} \\ \leq \int_{0}^{t} |G(\mathbf{r}_{m}, t; \mathbf{r})|^{2} d\tau \int_{0}^{t} |G(\mathbf{r}, t + \tau; \mathbf{r}_{m0})|^{2} d\tau$$
(4)

where $|\bullet|$ is the absolute value. Equation (4) can be rewritten as

$$|R_{TR}| \leq \left(\int_0^t |G(\mathbf{r}_m, t; \mathbf{r})|^2 d\tau\right)^{\frac{1}{2}}$$
$$\left(\int_0^t |G(\mathbf{r}, t+\tau; \mathbf{r}_{m0})|^2 d\tau\right)^{\frac{1}{2}}.$$
(5)

As the Euclidean norm and signal energy, respectively, are defined as

$$\|G(\mathbf{r}_m, t; \mathbf{r})\| = \left(\int_0^t |G(\mathbf{r}_m, t; \mathbf{r})|^2 dt\right)^{\frac{1}{2}}$$
(6)

$$E_{G\mathbf{r}_m} = \int_0^t |G(\mathbf{r}_m, t; \mathbf{r})|^2 dt$$
(7)

(De Simone et al., 2018). Hence, Equation (5) can be written as

$$|R_{TR}| \le \|G(\mathbf{r}_m, t; \mathbf{r})\| \|G(\mathbf{r}, t + \tau; \mathbf{r}_{m0})\| = \sqrt{E_{G\mathbf{r}_m} E_{G\mathbf{r}_{m0}}}.$$
 (8)

The TR correlation coefficient, C_{TR} , defined as

$$C_{TR} = max \left(\frac{|R_{TR}|}{\sqrt{E_{Gr_m} E_{Gr_{m0}}}} \right)$$
(9)

can be used as a measure of similarity between two signals (De Simone et al., 2018). Comparing Equations (8) and (9), it is clear that the following inequality is satisfied, $0 \leq C_{TR} \leq$ 1, where a C_{TR} value near to one represents strong signal similarity and identifies the impact location. The C_{TR} value is calculated for all baseline excitation points and those points nearest to the impact have the highest values. As reported in the Introduction section, although only one sensor is required for the imaging process, it has been shown that using multiple sensors and, then, averaging the C_{TR} values associated to each baseline point given by the readings from each sensor, allows for the compensation of incoherent measurement noise due to electronics, environmental, and operational conditions (Ciampa et al., 2016; Castro et al., 2019).

TOPOLOGICAL APPROACHES FOR IMPACT LOCALIZATION

In order to improve the accuracy of the impact location, a new topological approach, here named as the MA method is proposed, which resembles the traditional CoG technique. The effectiveness of this new method is discussed in the following sections. For all impact cases, the absolute error of the calculated impact locations was used, and it was defined as the Euclidian distance between the calculated and the real impact position:

$$\chi = \sqrt{(x - x_0)^2 + (y - y_0)^2}$$
(10)

where *x* and *y* are the calculated coordinate positions and x_0 and y_0 are the actual impact coordinates.

Traditional CoG Method

In the traditional CoG method (Chen et al., 2012b; Xu et al., 2016; De Simone et al., 2018), after a baseline grid of $M C_{TR}$ values are obtained, the average of the C_{TR} values relating to the four nodes of each grid cell is calculated. The resulting coefficient, is called the "global time reversal coefficient," C_{TR} *GLOBAL*. The cell with



the highest C_{TR_GLOBAL} value is then selected as the impact cell and its Cartesian coordinates are calculated using

$$x = \frac{\sum_{i=1}^{4} x_i C_{TRi}}{\sum_{i=1}^{4} C_{TRi}}, \quad y = \frac{\sum_{i=1}^{4} y_i C_{TRi}}{\sum_{i=1}^{4} C_{TRi}}$$
(11)

where *x* and *y* are coordinate positions relative to the bottom left corner of the test plate and subscripts i = 1-4 represent the four corners of the impact cell. Experimental analysis revealed that the CoG method always gives very little variation from the center of the predicted impact cell. Therefore, an alternative method to the traditional CoG technique was investigated in the following section.

Minimum Average (MA) Approach for Enhanced Impact Localization

In the new MA approach proposed in this work, the Cartesian coordinates for the estimated impact location are calculated differently using the minimum value of the TR correlation coefficient:

$$x = \frac{\sum_{i=1}^{4} x_i (C_{TRi} - \min(C_{TRi}))}{\sum_{i=1}^{4} (C_{TRi} - \min(C_{TRi}))},$$

$$y = \frac{\sum_{i=1}^{4} y_i (C_{TRi} - \min(C_{TRi}))}{\sum_{i=1}^{4} (C_{TRi} - \min(C_{TRi}))}$$
(12)

where the differences from the minimum C_{TR} value of the four corners of the impact cell are used rather than the actual C_{TR} values. This method is similar to the CoG technique, however, differences between C_{TR} values of the predicted cell corners provide additional flexibility to the localization process, thus permitting deviations from the geometric center of the impact cell. The scope of the MA approach was not to resolve the selection of the wrong cell, but to improve the accuracy of both off-cell center impacts and impact cases where the wrong cell is predicted.

EXPERIMENTAL SET-UP

All tests described in this paper were conducted on a single CFRP plate with dimensions of $200 \times 200 \times 3$ mm and a layup sequence of [90/0]3s. As can be seen in Figure 1, the plate contained four embedded PZT sensors of 6.5 mm diameter and 0.3 mm thickness, which were located between the fourth and fifth layers from the top, covered with thin $10 \times 10 \text{ mm}$ woven E-glass fiber fabric patches. The manufacturing process for the electrical insulation of PZT transducers was described in Andreades et al. (2018). Each sensor (labeled A-D) was positioned 20 mm from the edges in the four corners of the plate. A baseline grid of hundred 15×15 mm cells were marked onto the plate as shown in Figures 2A,B. Impacts were generated by dropping a pencil from a consistent height of 85 mm. To accurately impact desired positions on the plate, while releasing the pencil from the same height, a tube held by a retort stand and clamp was used. This set-up is shown in Figure 2A.

The plate was suspended with four foam squares, to prevent interactions and reflections with the table. The impact signal responses from PZT sensors were acquired using a four channel PicoScope, with 8 bits of resolution, at a sampling rate of 2 MHz and an acquisition time of 50 ms. A single threshold trigger was used with a 5% pre-trigger time set to ensure the whole impulse response was captured.

To validate the robustness of the proposed work, practical experimental conditions were applied on each impact test in order to alter AE responses. These included a refining of the





baseline grid (please see section Pencil Drop test and Grid Refinement) and the presence of material damage and double impacts, both causing variation of the baseline signals (please see section Double Bounce Impact and Baseline Variation With Damage).

Pencil Drop Test and Grid Refinement

The original baseline grid, in which each point pertains to a corresponding recorded impulse response, as shown in **Figure 2B**, relies on 121 points. In order to optimize the training process of the plate, a grid refinement factor of 2 was proposed.





no grid refinement.

In this case, the concept of grid refinement factor 2 means that the number of baseline points used is reduced by 2 in relation to the original grid. Such a training corresponds to the baseline cell sides being doubled, making the baseline grid coarser and reducing the number of excitation points. **Figure 3** shows the original and the refined baseline grid dimensions as well as the impact locations highlighted by the numbers. It can be observed that the impact points were increased from 100 to 125, and the number of baseline has decreased significantly from 121 to 36 points, by the grid refinement factor 2. After the training process, the original CoG method and the new topological MA approach were applied for a comparison of impact localization.

Double Bounce Impact and Baseline Variation With Damage

Before the evaluation of the double bouncing condition, the repeatability of the proposed TR imaging process was analyzed. Theoretically, two impacts at the same location should produce the same impulsive response. This repeatability was tested for five impact points at locations shown in Figure 4A. Following impact tests, a double bouncing test was achieved in the same experimental configuration. In this case, the pencil was dropped and left to settle, not pulled back after the first bounce, resulting in double impacts in quick succession. The response was correlated to the baseline grid. Furthermore, in real aerospace applications, it is possible that the structure may develop some form of damage, which may not affect the system structurally, but it may alter the resultant wave field generated from impacts. If the elastic response is altered too, then the original baseline data will need to be recaptured, which is time-consuming. In addition, if the damage is not detected or known then the system will give incorrect impact locations without the user realizing. Therefore, in order to test how sensitive the system was to damage, a 5 mm diameter hole was drilled into the center of the plate. Impact positions were selected to show a spread of responses around the plate both near and far from the hole, according to **Figure 4B**.

IMPACT LOCALIZATION RESULTS

Impact localization results are divided into two sub-sections. In the first one, the effectiveness of TR for impact localization with the new MA approach and embedded piezoelectric sensors were evaluated on the composite plate. Both the original and the refined grid were considered. In the second sub-section, impact localization results with double bouncing and damage test were reported.

Pencil Drop Test and Grid Refinement Results

The original baseline grid made of 15×15 mm cells was firstly evaluated. All impacts were aligned with the center of each cell.

TABLE 1 | Post-processing results for no grid refinement, CoG method and the new MA approach.

 TABLE 2 | Post processing results for grid refinement, CoG method and the new topological approach.

Method	Mean Error (mm)	Standard Deviation (mm)	Maximum C _{TR_GLOBAL}	Minimum C _{TR_GLOBAL}	
CoG method	0.25	0.14	0.86 (impact 50)	0.70 (impact 19)	
MA method	5.23	1.69	0.86 (impact 50)	0.70 (impact 19)	

Method	Mean error (mm)	Standard deviation (mm)	Maximum C _{TR_GLOBAL}	Minimum C _{TR_GLOBAL}
CoG method	14.90	10.11	0.65 (impact 114)	0.44 (impact 33)
MA method	10.70	7.48	0.65 (impact 114)	0.44 (impact 33)



FIGURE 6 | Calculated and actual impact locations for the CoG method (A) and the new topological MA approach (B) applied to TR with off-centered impacts and grid refinement.

Results for the CoG method and the MA approach are displayed in **Figures 5A,B**, respectively.

It can be verified that, as shown in the **Figure 5**, the traditional CoG method presented a very small variation from the center of the predicted cells, whereas the MA technique showed variations from the center. Hence, for cell-centered impacts with the CoG method, all correct impact cells were predicted, and the mean errors and standard deviations were low in relation to the MA approach. **Table 1** provides the statistical parameters and the values of C_{TR_GLOBAL} maximum and minimum for both cases. As reported in section Traditional CoG Method, each impact can be used to generate C_{TR_GLOBAL} values for

all cells and it is the cell with the maximum C_{TR_GLOBAL} value that becomes the predicted cell. The maximum and minimum C_{TR_GLOBAL} values in **Table 1** refer to the maximum and minimum C_{TR_GLOBAL} values associated to each predicted cell for all impacts. The maximum and minimum C_{TR_GLOBAL} values are useful to indicate the strength of the correlation between the impact and the recorded baseline grid, i.e., the higher the value the more likely the correct impact cell will be identified.

From Table 1, it can be observed that the mean error was increased from 0.25 to 5.23 mm, and the standard deviation increased from 0.14 to 1.69 mm, showing that the accuracy of



the traditional CoG methodology was higher. However, for offcenter impacts and larger cells the new topological MA method provided more accurate results, as observed in **Figure 6** and **Table 2**.

Results presented in **Figure 6** showed that all impacts were predicted around the center of the cell with the traditional CoG method, since this methodology does not ponder the weight of each C_{TR} value, as opposite to new topological approach. The mean error of the GoG technique was 14.9 mm against 10.7 mm of the MA method. For the standard deviation, calculated values were 10.11 mm for the GoG method and 7.48 mm for the MA technique. According to **Tables 1**, **2**, the grid refinement decreased both maximum and minimum values of the TR correlation coefficient, as the refinement provided less data amount in relation to the original grid. However, although the mean error and standard deviation increased in relation to the denser grid, the refinement can be an alternative for systems that require less data to perform the impact localization.

From **Figure 6B** it is also clear that the localization for border impacts have best estimation in relation to the cellcenter impacts. This results is due to the non-equilibrium of the C_{TR} of the corners of the coarse grid. If one point dominating, showing bigger C_{TR} values compared to the other three corners, the localization of cell-center impact lie off the correct point. To better understand this phenomenon C_{TR} plots of the maximum and minimum predicted cell C_{TR_GLOBAL} values are shown in **Figure 7**. According to the **Figures 7A,B**, despite the rough surface for this grid refinement, there are clear peaks that coincide with the impact locations, even for impact 19 where the predicted cells C_{TR_GLOBAL} value is the smallest of all impacts. In relation to the **Figures 7C,D**, it can be seen that the peaks are becoming less prominent, especially for the minimum C_{TR_GLOBAL} value, which corresponds to a cell-centered impact, as shown on **Figure 6A**, meaning the correlation to all points is reduced as fewer baseline points are as close to the actual impact location. For an impact at the corner of a cell, such as impact 114, it can be seen that one point correlates well-leading to the peak being more of a single point. This can introduce more risks as the predicted impact cell could fall incorrectly on the wrong side of this peak, especially if there is a secondary peak on that side. The new topological MA approach would compensate this error well, whereas the traditional CoG method would not. This is likely why the new MA approach revealed lower mean errors and standard deviations for the grid refinement.

Evaluation of the Repeatability of the TR Impact Localization Process

In this section, the repeatability of the proposed TR localization process with embedded PZT transducers was evaluated by altering the baseline signals with double bouncing tests and by introducing material damage (5 mm diameter hole) into the composite plate.

Double Bouncing Tests

Figure 8 shows the C_{TR} values of the repeatability and double bouncing test. Results illustrate that, in relation to the repeatability test, all impacts produced the same impulsive response, since all C_{TR} values tend to 1. In relation to the double







bouncing test, the response was only slightly affected and the resulting C_{TR} values from the correlation between the single impacts and the double impacts were all still near to one. The

impact point "1" presented the lowest C_{TR} values of the test, which were, 0.96, 0.92, and 0.96 for the sensor B, C, and D. For the other impact tests the index remained close to the unit.

TABLE 3 | Post-processing results for the original grid.

Plate condition	Method	Mean error (mm)	Standard deviation (mm)	Maximum C _{TR_GLOBAL}	Minimum C _{TR_GLOBAL}
No damage	CoG method	0.22	0.09	0.85 (impact 8)	0.79 (impact 1)
	MA method	5.09	1.42	0.85 (impact 8)	0.79 (impact 1)
Damage	CoG method	0.49	0.25	0.85 (impact 8)	0.79 (impact 1)
	MA method	5.77	0.65	0.85 (impact 8)	0.79 (impact 1)



with damage.

Damaged Plate

Figures 9, **10** show the damaged plate impact localization with no grid refinement and the refined grid, respectively. In order to compare the influence of failure the previously trained grids with no damage condition were here considered as a baseline. TR impact localization results revealed that impact localization

could be achieved also in the damaged plate. The CoG method presented better results in relation to the MA method for no grid refinement, as the mean error value and the standard deviation were lower. However, for the grid refinement condition the MA technique was more accurate, as the CoG approach tends to concentrate impacts in the center of the cell. These

TABLE 4 | Post processing results with grid refinement.

Plate condition	Method	Mean error	Standard	Maximum	Minimum
		(mm)	deviation (mm)	C _{TR_GLOBAL}	CTR_GLOBAL
No damage	CoG method	9.94	4.82	0.66 (impact 8)	0.60 (impact 6)
	MA method	7.34	3.55	0.66 (impact 8)	0.60 (impact 6)
Damage	CoG method	8.49	0.66	0.67 (impact 8)	0.56 (impact 6)
	MA method	6.62	0.73	0.67 (impact 8)	0.56 (impact 6)



without damage present. (A) No damage maximum C_{TR_GLOBAL} ; (B) no damage minimum C_{TR_GLOBAL} ; (C) same impact location as (A) with damage; (D) same impact location as (B) with damage.

results were also observed in the section Pencil Drop Test and Grid Refinement Results. The statistical analysis is summarized in **Table 3**.

It can be seen that the damage increases slightly the mean error, which, for the CoG reached 0.27 mm and for the MA technique raised up to 0.69 mm. The variation in C_{TR} distributions for impact locations relating to the maximum and

minimum C_{TR_GLOBAL} values with no damage are shown in **Figure 11** for both the cases with and without damage.

The location and magnitudes of peaks did not vary significantly with the presence of damage, however, the profile of the peaks did. This resulted in the same predicted impact cells being identified but variations with the predicted impact locations within them. This is particularly evident for the minimum C_{TR_GLOBAL} value impacts, the whole C_{TR} distribution was altered giving the secondary peak greater magnitude. This can significantly change predicted impact locations with coarser grids when both the primary and secondary peaks are encompassed by one baseline cell. **Table 4** presented the damage plate analysis for the grid refinement factor 2.

As reported in **Table 4**, although errors increased and all C_{TR_GLOBAL} indices decreased in relation to the first grid, the MA method was more effective to accomplish the impact localization with the refined grid in the damage plate. **Figure 12** shows the C_{TR} distributions for all conditions in the damaged plate.

According to **Figure 12**, the C_{TR} distributions did not vary greatly with the introduction of damage, even less so than that seen with the unrefined grid. This explains why for most impacts the variation in calculated impact locations varied less for the coarse grid than that seen for the original grid.

CONCLUSION

This study investigated the performance of time reversal for the localization of the impact sources on a fiber reinforced plastic composite with embedded piezoelectric sensors. A topologic approach was proposed in order to enhance the accuracy of time reversal in retrieving impact location. Experimental tests were accomplished to assess the effectiveness of time reversal against many practical issues in real conditions, such as material imperfections and double bouncing, , which were not consider

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during the initial training process. Based on the results presented, it can be concluded that the time reversal applied on a composite with embedded piezoelectric sensors proved to be an effective impact localization methodology where alterations of acoustic emission in the baseline signals are less predictable, once the impacts will be possibly distributed equally over the surface of the structure. In addition, the minimum average approach proved to be an alternative for systems that require less data to perform the impact localization. Future work will focus on the on reproducibility. This will include the use of multiple baselines and weighting functions for an improved selection of the cell under consideration for impact localization.

DATA AVAILABILITY STATEMENT

The raw data supporting the conclusions of this article will be made available by the authors, without undue reservation, to any qualified researcher.

AUTHOR CONTRIBUTIONS

AC and BC run the time reversal experiments. FB supported the development of ultrasound testing. CA manufactured the composite plate with embedded transducers. MM supported the development of time reversal theory. FC supported the development of time reversal for impact localization and led the research team.

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Conflict of Interest: The authors declare that the research was conducted in the absence of any commercial or financial relationships that could be construed as a potential conflict of interest.

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